

1 A CFD-sectional algorithm for population balance equation coupled with multi-dimensional flow  
2 dynamics

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8 ABSTRACT

9 The sectional methodology has become a powerful tool for population balance modeling.  
10 However, in coupling with the computational fluid dynamics (CFD), there exist several  
11 challenges, including the complex double integrals associated with the sectional coefficients,  
12 the serious computational burden in the solution of a large number of partial differential  
13 equations, and the nonlinear and coupled source terms. In the present study, a novel CFD-  
14 sectional algorithm that aims at addressing these challenges is developed in the open-source  
15 package – OpenFOAM to solve population balance equation coupled with multi-dimensional  
16 flow dynamics. The sectional coefficients are specified by a numerical quadrature with adaptive  
17 adjustment of integration limits. The inter-equation coupling is treated by hybrid-segregated  
18 procedures and the source term is formally linearized by the operator splitting method. The  
19 numerical quadrature proves to be computationally efficient and accurate in solving the  
20 sectional coefficients. The developed CFD-sectional algorithm is validated against a self-  
21 preserving solution of particles undergoing Brownian coagulation. The acoustic agglomeration  
22 in a standing wave is simulated as a representative case study. It has been demonstrated that  
23 the predictions regarding the particle size distribution and agglomeration process agree well  
24 with the experimental data, which verifies the capability of the present CFD-sectional algorithm  
25 in simulating the spatially inhomogeneous population balance equation coupled with multi-  
26 dimensional flows.

27

28 Keywords: population balance modeling; sectional method; computational fluid dynamics (CFD);  
29 coalescence/agglomeration; acoustic agglomeration

30

## 31 1. Introduction

32 Population balance modeling is of great importance in a wide spectrum of particulate  
33 processes, ranging from combustion, spraying, nano-powder synthesis, to granulation etc. [1-7].  
34 Although the population balance equation (PBE) is dependent on both time and space in  
35 nature, numerical simulations of PBE traditionally assume that the aerosol is homogeneous  
36 within the geometric boundaries, indicating that all the properties of the aerosol are identical  
37 everywhere without spatial variation of any quantity involved in the process [8]. Under such  
38 circumstances, the PBE can only capture the temporal evolution of the particle size distribution  
39 (PSD), which is the so-called temporal model. The temporal PBE is a partial integro-differential  
40 equation and has only a limited number of known analytical solutions. As such, approximate  
41 solutions to the temporal PBE are often determined by using several numerical approaches,  
42 which can be classified as the method of moments [3, 4, 9-11], Monte Carlo method [12-14],  
43 the sectional method [15-17] and the fixed pivot method or cell average method [18-20]. The  
44 temporal model can explore some basic characteristics of the particulate system, however, its  
45 application is restricted by the homogeneity assumption to a limited number of scenarios since  
46 most of aerosol systems in industrial engineering or laboratory experiments are spatially  
47 inhomogeneous, i.e., multi-dimensional. In addition, the temporal PBE does not take into  
48 account any coupling of the aerosol processes with the fluid flow or heat transfer, which limits  
49 its ability to achieve the complete information of particulate laden flows. Therefore, the tempo-  
50 spatial formulation of the PBE must be solved by coupling with a computational fluid dynamics  
51 (CFD) framework to reveal the essential features of aerosol dynamics in a realistic multi-  
52 dimensional flow.

53 In the CFD-based population balance modeling, all the physical variables involved need to be  
54 discretized in three spatial directions in addition to the temporal and internal coordinates. This  
55 greatly increases the complexity of the algorithm and thus, compared to the extensive  
56 numerical simulations of the temporal model, only a limited number of studies have been  
57 reported to solve the spatially multi-dimensional population balance equation in the presence  
58 of fluid flows. Stratmann and Whitby [21] pioneered a preliminary numerical solution for the  
59 combined CFD-population balance modeling of aerosol dynamics by extending computational

60 heat transfer and fluid flow codes. Kruis et al. [22] simplified the population balance equation  
61 based on a mean particle size, and developed a monodisperse population balance (MPB)  
62 model. Buddhiraju and Runkana [23] simulated the nanoparticle synthesis in a flame reactor by  
63 the CFD-MPB model, and their prediction was basically in agreement with the experimental  
64 data. In spite of the robustness, simplicity and computation efficiency, the MPB model fails to  
65 give sufficient insights to the particle characteristics and deteriorates the simulation quality for  
66 polydisperse aerosols [24]. In general, the population balance equations by using the method of  
67 moments (MOM) are not closed when modeling the size-dependent aerosol process with  
68 coagulation or breakage [9]. Several approaches are introduced to solve the closure problem, of  
69 which the most common includes assuming a specific functional form of the particle size  
70 distribution *a priori*, and quadrature approximation such as quadrature method of moments  
71 (QMOM). Schwade and Roth [25] solved the tempo-spatial PBE for aerosol dynamics via the  
72 moment methodology based on the assumption of a unimodal lognormal size distribution.  
73 However, in most engineering applications, the particle size distribution of an aerosol may not  
74 be known beforehand and its functional form may be changed with time or space. Thus, the  
75 QMOM is more versatile and promising in the context of moment based methods. Passalacqua  
76 et al. [26] constructed a PBE solver based on the extended quadrature method of moments  
77 (EQMOM) and then coupled it to a standard CFD code package in OpenFOAM, which enables  
78 the solution of the PBE for inhomogeneous systems with transport of particles due to the fluid  
79 motion. Although the moment methodology presents the advantage of being less demanding  
80 on computational resources, it is plagued by the assumption of a specific particle size  
81 distribution or the inversion problem that arises in the reconstruction of the distribution details  
82 from the moment integrals. In addition, it is reported that the realizability cannot be  
83 guaranteed for a scheme higher than second-order based on the reconstruction of moments  
84 [27]. Another method to resolve the PBE is through the use of the stochastic method or Monte  
85 Carlo (MC) method. Liu and Chan [13] developed a coupled CFD-Monte Carlo method to  
86 simulate the complex aerosol dynamics in turbulent flows at different Reynolds numbers. The  
87 major limitation of this method is the extremely high computation cost since the time step is  
88 required to be small, and the tracking of individual particles is computationally complicated.

89 Fan et al. [28] adopted the MC method to simulate a typical case of acoustic agglomeration in a  
90 small physical space with only 7,650 particles but the computation time was up to 4 days on a  
91 computer equipped with 4 CPUs (Intel i5) and RAM of 8192 MB.

92 The sectional method, first proposed by Gelbard and Seinfeld [15], has been extensively used in  
93 the simulation of particulate systems. This method divides the continuous particle size  
94 distribution into a finite number of discrete sections, and within each section the particle  
95 distribution can be regarded as uniform or of any pre-defined function. The sectional method is  
96 advantageous due to the visual representation of the particle distribution and direct  
97 comparison with the experimental measurements. Plus, implementation of the sectional  
98 method in an Eulerian computational fluid dynamics framework is conceptually  
99 straightforward. Pyykonene and Jokineimi [29] constructed two CFD-based sectional schemes  
100 of population balance modeling coupled with fluid flows, one based on boundary layer type  
101 streamtube methodology and the other on full Eulerian representation. However, their full  
102 Eulerian sectional scheme failed to provide accurate predictions due to excessive demands on  
103 grid resolution. Jeong and Choi [30, 31] analyzed the growth of non-spherical polydisperse  
104 particles in a two-dimensional geometry by solving two sets of sectional PBEs coupled with  
105 detailed fluid dynamics calculations. Spatial transport phenomena such as convection, diffusion,  
106 thermophoresis and deposition were accounted for and comparisons have been made between  
107 the calculations using spatially two-dimensional approach and simplified one-dimensional  
108 approach, and the experimental data. It has been demonstrated that at high flow rates, the  
109 simplified one-dimensional approach would result in significant errors in comparison with the  
110 two-dimensional approach. Frederix et al. [32] applied the characteristics-based sectional  
111 method to the complete set of fluid equations for a compressible flow by incorporating the  
112 fractional step method. However, the feasibility of the developed algorithm was only  
113 demonstrated in studying the aerosol nucleation and condensation and did not cover the  
114 coagulation/agglomeration process. In addition, the variant of the sectional method, discrete-  
115 sectional method, has also been widely used in combination with CFD for the chemical  
116 synthesis of monomers and subsequent Brownian coagulation [33, 34].

117 Although the sectional methodology has become a powerful technique for population balance  
118 modeling, there exist several challenges or limitations in the current CFD-sectional algorithm: a)  
119 The sectional coefficients have to be solved via numerical quadrature of complex integrals  
120 while accurate evaluation of these integrals is computationally demanding, especially for the  
121 particles with a wide range of size spectrum. b) To obtain the refined particle size distribution,  
122 one has to solve a vast number of partial differential equations (PDEs). c) The source terms in  
123 the sectional PBE associated with the coagulation or agglomeration process are strongly  
124 nonlinear and coupled. As such, when conceiving the CFD-sectional algorithm, any  
125 inappropriate treatment can undermine the convergence and stability of the solution,  
126 especially for the ultrafast coagulation/agglomeration such as acoustic agglomeration. In view  
127 of the above challenges and limitations, the CFD-sectional method for spatially inhomogeneous  
128 population balance modeling in inhomogeneous flows remains to be an attractive topic.

129 In the present study, we develop a novel CFD-sectional algorithm for univariate population  
130 balance modeling in a multi-dimensional laminar or turbulent flow, targeting at settling the  
131 challenges in the existing sectional simulations. The sectional formulation of the PBE, which  
132 incorporates the convective and diffusive transport of aerosols, is constructed based on the  
133 sectional method of Gelbard and Seinfeld [15] and the coagulation/agglomeration kernel is  
134 present as an additional source term. A modified numerical integration scheme is proposed to  
135 improve the computation efficiency and accuracy in the calculation of the sectional coefficients.  
136 Subsequently, the framework of population balance modeling is described with focus on the  
137 processing of the inter-equation coupling and nonlinearity of the source terms. The sectional  
138 modeling of PBE is incorporated to a standard CFD algorithm in the open-source code library –  
139 OpenFOAM. The analytical solution of a self-preserving Brownian coagulation is chosen to  
140 validate the developed computer codes. A case study is carried out to simulate the acoustic  
141 agglomeration in a standing wave field and the numerical results are compared with  
142 experimental data to demonstrate the capability of the present CFD-sectional algorithm.

## 143 2. Sectional formulation of the population balance equation

144 The tempo-spatial evolution of particle size distribution for aerosols with coagulation or  
145 agglomeration process is governed by the population balance equation [35],

146 
$$\frac{\partial n(\vec{x}, v, t)}{\partial t} + \nabla \cdot [\vec{U}n(\vec{x}, v, t)] = \nabla \cdot [D\nabla n(\vec{x}, v, t)] + S \quad (1)$$

147 where, the particle size distribution function  $n(\vec{x}, v, t)$  is dependent on the space vector  $\vec{x}$ , time  
148  $t$  and particle volume  $v$ .  $U$  is the velocity of particles due to convective transport and  $D$  is the  
149 diffusion coefficient or diffusivity.

150 The source term  $S$  describes the net increase of aerosol particles due to the  
151 coagulation/agglomeration process [35],

152 
$$S = \frac{1}{2} \int_0^v \beta(\hat{v}, v - \hat{v}) n(\vec{x}, \hat{v}) n(\vec{x}, v - \hat{v}) d\hat{v} - \int_0^\infty \beta(v, \hat{v}) n(\vec{x}, v) n(\vec{x}, \hat{v}) d\hat{v} \quad (2)$$

153 where  $\beta$  is the kernel or frequency function in the coagulation/agglomeration process.

154 Following the notation of Gelbard and Seinfeld [15], a general property of an aerosol is  
155 introduced as

156 
$$q(\vec{x}, v, t) = \alpha v^\gamma n(\vec{x}, v, t) \quad (3)$$

157 where,  $\alpha$  and  $\gamma$  are constants that determine the general property of an aerosol (number,  
158 volume or surface area concentration etc.).

159 Dividing the entire particle size domain into a finite number of sections, one can define a  
160 discrete quantity of the aerosol in the  $l$ -th section as

161 
$$Q_l = \int_{v_{l-1}}^{v_l} q(\vec{x}, v, t) dv = \int_{v_{l-1}}^{v_l} \alpha v^\gamma n(\vec{x}, v, t) dv \quad (4)$$

162 where, for  $\alpha=1$  and  $\gamma=0$ ,  $Q_l$  is the particle number concentration  $N_l$ , while for  $\alpha=1$ ,  $\gamma=1$ ,  $Q_l$  is the  
163 particle volume concentration,  $V_l$ .

164 By applying Eq. (4) to the population balance equation Eq. (1), the sectional formulation for a  
165 general quantity of the aerosol in the  $l$ -th section can be written as follows

166 
$$\frac{\partial Q_l}{\partial t} + \nabla \cdot (\vec{U}Q_l) = \nabla \cdot (\vec{D}_l \nabla Q_l) + S_l \quad (5)$$

167 The sectional formulation of the source term in the  $l$ -th section,  $S_l$ , which results from the  
168 coagulation/agglomeration process, is written as

169 
$$S_l = \frac{1}{2} \sum_{i=1}^{l-1} \sum_{j=1}^{l-1} {}^1\beta_{i,j,l} Q_i Q_j - \sum_{i=1}^{l-1} {}^2\beta_{i,l} Q_i Q_l - \frac{1}{2} {}^3\beta_{l,l} Q_l^2 - \sum_{i=l+1}^{NS} {}^4\beta_{i,l} Q_i Q_l \quad (6)$$

170 where  $NS$  is the number of sections, and the sectional coefficients,  ${}^1\beta_{i,j,l}$ ,  ${}^2\beta_{i,l}$ ,  ${}^3\beta_{i,l}$  and  ${}^4\beta_{i,l}$ ,  
 171 represent the inter- and intra-sectional coagulation/agglomeration rates between sections prior  
 172 to the  $l$ -th section, between the  $l$ -th section and lower sections, within the  $l$ -th section and  
 173 between the  $l$ -th section and higher sections, respectively.

174 Often, the aerosol size distribution is not restrictively described by the particle volume  $u$  or  $v$   
 175 but the size variable  $\xi$  or  $\zeta$ , which is a function of the particle volume,  $\xi = f(u)$  or  $\zeta = f(v)$ . In  
 176 this situation, the particle size distribution is sectionalized on the basis of the size variable  $\xi$  or  $\zeta$   
 177 and accordingly the sectional coefficients are solved on  $\xi$  or  $\zeta$ . The detailed formulas of the  
 178 sectional coefficients are demonstrated in Table 1, which can also be referred to [15].

179 Table 1. Formulas of the sectional coefficients,  $\zeta_i = f(v_i)$ ,  $u = f^{-1}(\xi)$  and  $v = f^{-1}(\zeta)$ .

${}^1\beta_{i,j,l}$	$\int_{\zeta_{i-1}}^{\zeta_i} \int_{\zeta_{j-1}}^{\zeta_j} \frac{\theta(v_{l-1} < u + v < v_l)(u + v)^\gamma \beta(u, v, \vec{x})}{\alpha u^\gamma v^\gamma (\zeta_i - \zeta_{i-1})(\zeta_j - \zeta_{j-1})} d\xi d\zeta$
${}^2\beta_{i,l}$	$\int_{\zeta_{i-1}}^{\zeta_i} \int_{\zeta_{l-1}}^{\zeta_l} \frac{[\theta(u + v > v_l)u^\gamma - \theta(u + v < v_l)((u + v)^\gamma - u^\gamma)]\beta(u, v, \vec{x})}{\alpha u^\gamma v^\gamma (\zeta_i - \zeta_{i-1})(\zeta_l - \zeta_{l-1})} d\xi d\zeta$
${}^3\beta_{i,l}$	$\int_{\zeta_{l-1}}^{\zeta_l} \int_{\zeta_{l-1}}^{\zeta_l} \frac{[\theta(u + v > v_l)(u^\gamma + v^\gamma) + \theta(u + v < v_l)((u^\gamma + v^\gamma) - (u + v)^\gamma)]\beta(u, v, \vec{x})}{\alpha u^\gamma v^\gamma (\zeta_l - \zeta_{l-1})(\zeta_l - \zeta_{l-1})} d\xi d\zeta$
${}^4\beta_{i,l}$	$\int_{\zeta_{i-1}}^{\zeta_i} \int_{\zeta_{l-1}}^{\zeta_l} \frac{u^\gamma \beta(u, v, \vec{x})}{\alpha u^\gamma v^\gamma (\zeta_i - \zeta_{i-1})(\zeta_l - \zeta_{l-1})} d\xi d\zeta$

180  
 181 Many of the general moments of a particle size distribution appear in the expressions for the  
 182 integral properties of a particulate system. Four moments correspond to four common  
 183 properties of an aerosol. The zeroth, second, third and sixth moment are proportional to the  
 184 total number, total surface area, total volume and total volume square per unit space,  
 185 respectively. Therefore, the corresponding set of sectional equations conserving the zeroth,  
 186 second, third and sixth moment is called the  $n$ -model,  $s$ -model,  $v$ -model and  $v^2$ -model,  
 187 respectively. The sectional method based on Gelbard and Seinfeld [15] can conserve one  
 188 moment or property provided that the corresponding model is solved and it is straightforward  
 189 to conserve multiple moments via solving several models simultaneously. Adams and Seinfeld

190 [36] proposed the two-moment aerosol sectional (TOMAS) model which can conserve the total  
191 number as well as total volume. Kajino et al. [37] extended it to a triple-moment sectional  
192 (TMS) model by incorporating the total surface area. However, by this way, the number of the  
193 sectional equations to be solved will be doubled or tripled, which is very demanding on  
194 computation capacity. Thus, the present CFD-sectional solver does not incorporate the  
195 multiple-moment model, and the other moments or properties can be derived based on the  
196 conserved moment or property [38]. Wu and Biswas [39] reported that for the derived  
197 moments, the deviation caused by internal inconsistency is not significant if small values of  
198 sectioning space can be adopted, and they recommended the  $v$ -model with better performance  
199 of deriving other properties. Vanni [40] made similar conclusion via the  $v$ -model in terms of the  
200 comparative analysis of the number averaged particle size and the volume/mass averaged  
201 particle size. Therefore, the present CFD-sectional solver selects the  $v$ -model which satisfies the  
202 volume/mass conservation.

### 203 3. Numerical solution of the sectional population balance equation

204 Different from the general scalar transport equation, the sectional PBE requires special  
205 attention to challenges involved in the evaluation of sectional coefficients, discretization of  
206 both the external and internal coordinates as well as the solution process. In this part, we will  
207 firstly introduce a modified numerical integration scheme with adaptive integration limits which  
208 can evaluate the sectional coefficients accurately and efficiently. Then the framework of  
209 discretizing and solving the sectional PBE will be described with focus on the treatment of the  
210 inter-equation coupling and nonlinearities lying in the source terms.

#### 211 3.1. Specification of sectional coefficients

212 The sectional coefficients in Table 1 represent the probabilities that a pair of sections, such as  
213 the  $i$ -th and  $j$ -th section, collide with each other and form agglomerates lying within or out of a  
214 given “goal” particle size region, which is specified by the discontinuous function  $\vartheta$ . Gelbard and  
215 Seinfeld [15] proposed a numerical scheme for geometric sectionalization under which the  
216 upper limit of a section is at least twice its lower limit, i.e.,  $v_{i+1} \geq 2v_i$ . However, their scheme  
217 involves very complex formulas to eliminate the discontinuity from the integrands, and even

218 the dummy variables are introduced into the integration limits. This substantially complicates  
219 the solution of the sectional coefficients via the numerical integration. Herein, a numerical  
220 integration scheme modified by the adaptive adjustment of integration limits is presented for  
221 geometric sectionalization which has high accuracy and efficiency.

222 For the integrals in Table 1, the integration domain with original upper and lower limits is a  
223 rectangular region determined by the section boundary in the  $u$ - $v$  coordinate system. However,  
224 the “goal” region which is meaningful in the integration is irregular due to the discontinuous  
225 functions. As shown in Fig. 1, the “goal” region of the numerical integration for  ${}^1\beta_{i,j,l}$  is actually  
226 the trapezoidal black area which is bounded by the straight lines  $u + v = v_{l-1}$  and  $u + v = v_l$ .  
227 In the case that the difference between the sizes of two colliding particles is very large, the two  
228 sections which the two particles belong to will go far away from each other. In fact, for a  
229 particle which is ten times larger than the other one in diameter, the volume is a thousand  
230 times apart. Consequently, the integration domain is seriously skewed so that the “meaningful”  
231 “goal” region is insignificant compared to the whole integration domain. As such, for the  
232 numerical integration with original integration limits, most of the interpolating nodes make no  
233 contributions with very coarse nodes in the “goal” region. Therefore, a standard numerical  
234 quadrature may cause serious computation errors and cannot be applied directly.

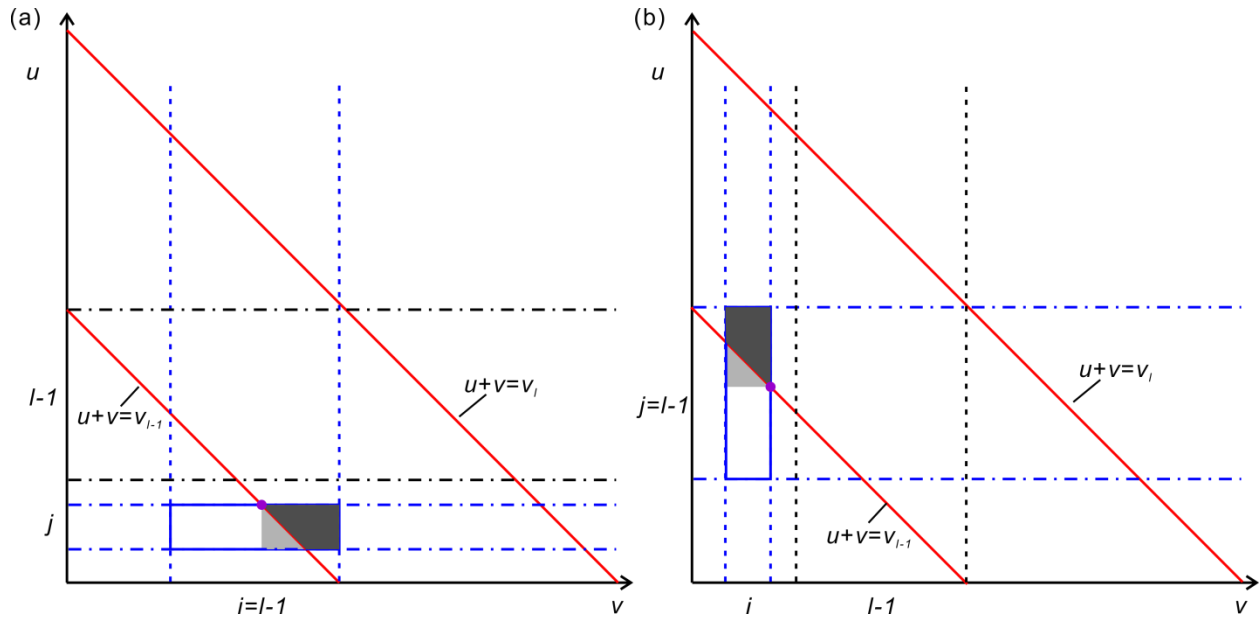
235 The geometric constraint implies that at least one of the colliding particles which forms an  
236 agglomerate in the  $l$ -th section must be in the  $(l-1)$ -th section, which means that  ${}^1\beta_{i,j,l} = 0$  for  
237 the section pairs of  $i < l - 1, j < l - 1$ . Since  ${}^1\beta_{i,j,l}$  represents the “birth” rate between  
238 sections prior to the  $l$ -th section, i.e.,  $i \leq l - 1, j \leq l - 1$ , the condition  $u + v < v_l$  is  
239 automatically satisfied. The other condition  $u + v > v_{l-1}$  determines the lower limit of  $v$   
240 coordinate for the case of  $i = l - 1, j < l - 1$ , or the lower limit of  $u$  coordinate for the case of  
241  $i < l - 1, j = l - 1$ . As shown in Fig. 1a for the case of  $i = l - 1, j < l - 1$ , in the presence of  
242 the discontinuous function, the lower limit of  $v$  coordinate cannot be smaller than the  
243 intersection point denoted by the purple circle marker while the  $u$  coordinate will not change.  
244 Hence, the integration limits can be adjusted adaptively according to the intersection point  
245 determined by the discontinuous function. To include the case of  $i = l - 1, j = l - 1$ , the

246 integration limits of  ${}^1\beta_{i,j,l}$  can be adjusted to  $[\max(v_{l-1} - v_j, v_{i-1}), v_i]$  for  $v$  coordinate and  
 247 left unchanged for  $u$  coordinate. Under such situation, most of the “meaningless” region is  
 248 eliminated from the original integration domain and the integration domain is modified as the  
 249 sum of the black “goal” region and the grey triangle region. For the same number of  
 250 interpolating nodes, this modified numerical integration scheme can increase the resolutions in  
 251 the “goal” region, which avoids a waste of the computation resources and greatly improves the  
 252 accuracy of the numerical integration. For the case of  $i < l - 1, j = l - 1$ , the limits of  
 253 integration become  $[\max(v_{l-1} - v_i, v_{j-1}), v_j]$  for  $u$  coordinate and remain unchanged for  $v$   
 254 coordinate. Similarly, the integration limits for  ${}^2\beta_{i,l}$  and  ${}^3\beta_{l,l}$  can also be adaptively adjusted  
 255 according to the intersection points as shown in Fig. 2 and Fig. 3. In the calculation of  ${}^2\beta_{i,l}$  the  
 256 two terms of the integrands represented by two discontinuous functions are split into two  
 257 integrals; when calculating  ${}^3\beta_{l,l}$ , the term with the discontinuous function,  $u + v > v_l$ , is  
 258 obtained by its complementary integral described by  $u + v < v_l$ . The results are tabulated in  
 259 Table 2 for programming basis.

260 Table 2. The integration limits with adaptive adjustment for the modified numerical integration scheme.

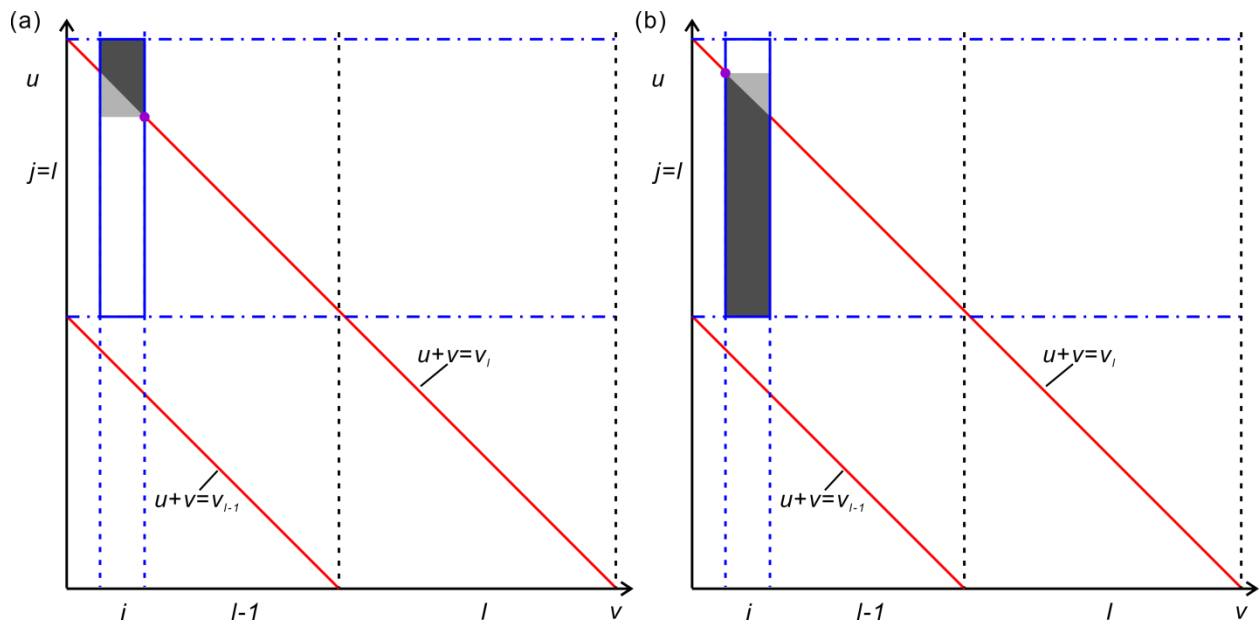
		Discontinuous Function	New Integration Limits	
			$u$ coordinate	$v$ coordinate
${}^1\beta_{i,j,l}$	$i = l - 1$	$\theta(v_{l-1} < u + v < v_l)$	$(v_{j-1}, v_j)$	$[\max(v_{l-1} - v_j, v_{i-1}), v_i]$
	$j = l - 1$	$\theta(v_{l-1} < u + v < v_l)$	$[\max(v_{l-1} - v_i, v_{j-1}), v_j]$	$(v_{i-1}, v_i)$
${}^2\beta_{i,l}$		$\theta(u + v > v_l)$	$(v_l - v_i, v_l)$	$(v_{i-1}, v_i)$
		$\theta(u + v < v_l)$	$(v_{l-1}, v_l - v_{i-1})$	$(v_{i-1}, v_i)$
${}^3\beta_{l,l}$		$\theta(u + v < v_l)$	$(v_{l-1}, v_l - v_{l-1})$	$(v_{l-1}, v_l - v_{l-1})$
		$\theta(u + v > v_l)$	$1 - \theta(u + v < v_l)$	
${}^4\beta_{i,l}$		N.A.	N.A.	N.A.

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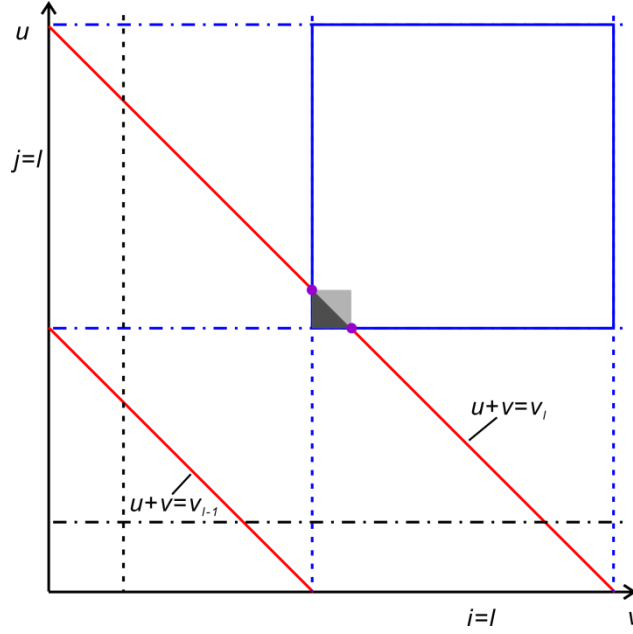
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Fig. 1. Illustration of the integration limits of  ${}^1\beta_{i,j,l}$  for the case of: (a)  $i = l - 1, j < l - 1$  and (b)  $i < l - 1, j = l - 1$ . The black shaded area shows the “goal” region and  $v_{l-1}$  and  $v_l$  represent the lower and upper boundaries of the agglomerating section.



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Fig. 2. Illustration of the integration limits of  ${}^2\beta_{i,l}$  for the case of: (a)  $\theta(u + v > v_l)$  and (b)  $\theta(v_{l-1} < u + v < v_l)$ . The black shaded area shows the “goal” region and  $v_{l-1}$  and  $v_l$  represent the lower and upper boundaries of the agglomerating section.



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Fig. 3. Illustration of the integration limits of  ${}^3\beta_{i,l}$ . The black shaded area shows the “goal” region and  $v_{i-1}$  and  $v_i$  represent the lower and upper boundaries of the agglomerating section.

282 The sectional coefficients can be solved by some kind of numerical quadrature such as  
283 composite Newton-Cotes quadrature modified by the adaptive integration limits in Table 2. The  
284 numerical integration formulation of a sectional coefficient for Newton-Cotes quadrature can  
285 be represented by

$$I = \iint_R F(u, v) du dv = \sum \sum \lambda_{ij} F_{ij} \quad (7)$$

287 where  $F(u, v)$  is the discontinuous integrand,  $R$  is the integration domain after being adaptively  
288 adjusted,  $\lambda_{ij}$  is a matrix of coefficients corresponding to the quadrature formula such as  
289 Trapezoid rule or Simpson’s rule and  $F_{ij}$  is the value at the interpolating node which is calculated  
290 by

$$F_{ij} = \begin{cases} \beta(u_j, v_i), & \theta \text{ function is satisfied} \\ 0, & \text{others} \end{cases} \quad (8)$$

292 The proposed numerical integration scheme with the adaptive adjustment of integration limits  
293 can achieve high accuracy and economic consumption of computation resource. In the  
294 meantime the complexity of this scheme is greatly alleviated compared to that of Gelbard and  
295 Seinfeld [15] and the programming for the numerical integration is very straightforward.

296 3.2. Convection term

297 We must specify the local velocity components of a non-uniform flow field before proceeding  
298 to discretize the convection term of the sectional population balance equation. Thus, the  
299 solution procedure of the sectional PBE should be embedded into a CFD program such as the  
300 SIMPLE algorithm proposed by Patankar [41] for steady-state problems or PISO algorithm  
301 proposed by Issa [42] for transient problems. The SIMPLE or PISO scheme consists of a repeated  
302 loop of momentum predictor, pressure solution, and velocity correction and the details can be  
303 referred to [41, 42]. With the conservative fluxes or velocity fields, the convection term in  
304 sectional PBE is linear and thus the discretization does not present particular problems when  
305 the traditional Gauss upwind is used. We will describe how to incorporate the solution  
306 procedure of sectional PBE into SIMPLE or PISO loop in Section 4.

307 3.3. Diffusion term

308 In a laminar particle laden flow, the diffusion of particles results from the random Brownian  
309 motion and this is called Brownian diffusion. However, in addition to the Brownian motion, the  
310 particles in a turbulent flow also move with the irregular turbulent fluctuations, which is the so-  
311 called turbulent diffusion. It has been shown that Fick's law can be applied to turbulent  
312 diffusion just as it does to Brownian diffusion [43]. Hence, in a turbulent flow the diffusion term  
313 can be modeled simply by the regular formulation  $\nabla \cdot (D\nabla n)$  except that the diffusivity  $D$  is the  
314 sum of Brownian diffusivity  $D_B$  and turbulent diffusivity  $D_T$ , i.e.,  $D = D_B + D_T$ .

315 The Brownian diffusivity,  $D_B$ , is calculated by the well-known Stokes-Einstein equation with  
316 Cunningham coefficient [44],

317 
$$D_B(d_p) = \frac{k_B T C_C}{3\pi\mu d_p} \quad (9)$$

318 where the Cunningham coefficient  $C_C$  is calculated by

319 
$$C_C = 1 + Kn[A + B\exp(-C/Kn)] \quad (10)$$

320 where,  $A$ ,  $B$  and  $C$  are three constants,  $A=1.257$ ,  $B=0.4$  and  $C=1.1$ , and the Knudsen number  
321  $Kn=2\lambda/d_p$ , and  $\lambda$  is the free path of air molecules. In the continuum regime, the Cunningham  
322 coefficient is equal to unity.

323 Depending on the flow conditions such as the Reynolds number ( $Re$ ), the turbulent diffusion of  
 324 particles can be several orders of magnitude more intense than Brownian diffusion. In turbulent  
 325 flows, particles are usually caused by their inertia to lag behind or to shoot ahead of the local  
 326 fluid motion (at random). However, on the average, these tendencies cancel out, which means  
 327 that the root mean square (RMS) fluctuating velocity of particles is equal to the turbulent  
 328 velocity of the local fluid. Thus the turbulent diffusivity of the particles can be assumed to equal  
 329 the local turbulent or eddy viscosity, which means that  $D_T = \nu_t$ , where,  $\nu_t$  is the turbulent or  
 330 eddy viscosity [43].

331 For the Reynolds Averaged Simulation (RAS) turbulence modeling, the turbulent viscosity can  
 332 be calculated according to the turbulent characteristics,

$$333 \quad \nu_t = C_\mu \frac{k^2}{\varepsilon} \quad (11)$$

334 where,  $C_\mu$  is a constant,  $C_\mu = 0.09$ ,  $k$  the turbulent kinetic energy and  $\varepsilon$  energy dissipation rate.

335 Since the diffusivity  $D$  is dependent on the particle size, the sectional formulation of the  
 336 diffusion term in the population balance equation is, in fact, described by,

$$337 \quad \int_{v_{l-1}}^{v_l} \alpha v^\gamma \nabla \cdot [D(v) \nabla n] dv = \nabla \cdot \int_{v_{l-1}}^{v_l} \alpha v^\gamma [D(v) \nabla n] dv \quad (12)$$

338 To decouple the integrand in Eq. (12), the sectional diffusivity is regarded as a constant in each  
 339 section, e.g., its average value,

$$340 \quad \bar{D}_l = \frac{1}{v_l - v_{l-1}} \int_{v_{l-1}}^{v_l} D(v) dv \quad (13)$$

341 Then, the diffusion term can be sectionalized as

$$342 \quad \int_{v_{l-1}}^{v_l} \alpha v^\gamma \nabla \cdot [D(v) \nabla n] dv = \nabla \cdot \left[ \bar{D}_l \int_{v_{l-1}}^{v_l} \alpha v^\gamma \nabla n dv \right] = \nabla \cdot (\bar{D}_l \nabla Q_l) \quad (14)$$

343 which is exactly the first term on the right hand side in the sectional population balance  
 344 equation. As a typical Laplacian term, the sectional diffusion term Eq. (14) is usually discretized  
 345 by the Gauss central differencing scheme.

#### 346 3.4. Source term

347 Let us return to Eqs. (5) and (6) for the sectional population balance equation. It can be seen  
 348 that the source term of the PDE for the  $l$ -th section is a function of all the dependent variables,  
 349  $Q_1 \cdots Q_{NS}$ , and highly coupled with the PDEs in the other sections. Thus, the solution of each  
 350 unknown involves all the other dependent variables. However, the particle size range of  
 351 interest is usually divided into tens of sections and each section generates one unknown. Thus,  
 352 a simultaneous solution of all unknowns is not preferred due to the tremendous cost of  
 353 computation resource. When we solve the coupled PDEs in sequence, the inter-equation  
 354 coupling associated with the source terms must be handled at first.

355 A simple method to decouple the source term is that we substitute “old” values of the initial  
 356 guess or in the previous loop for all the other variables except  $Q_l$ ,

$$357 \quad S_l = \frac{1}{2} \sum_{i=1}^{l-1} \sum_{j=1}^{l-1} {}^1\beta_{i,j,l} Q_i^o Q_j^o - \sum_{i=1}^{l-1} {}^2\beta_{i,l} Q_i^o Q_l - \frac{1}{2} {}^3\beta_{l,l} Q_l^2 - \sum_{i=l+1}^{NS} {}^4\beta_{i,l} Q_i^o Q_l \quad (15)$$

358 where, a superscript “o” denotes the “old” values. In this manner, the solution of the PDEs is  
 359 fully segregated in a SIMPLE or PISO loop, which means that the inter-equation couplings  
 360 among the variables  $Q_1 \cdots Q_{NS}$  would always be lagged in the solution process. This lagging  
 361 effect will, of course, reduce the speed of convergence and also bring about an error source,  
 362 which will be serious in transient flows due to the error accumulation with the time marching.

363 Herein, in the calculation of the  $l$ -th section,  $Q_l$ , the other variables involved are treated in a  
 364 hybrid-segregated way: the variables,  $Q_1 \cdots Q_{l-1}$ , adopt the “new” values which have been  
 365 updated in the current loop while the other unknown variables adopt the “old” values in the  
 366 last loop. Such a hybrid-segregated computation scheme can be expressed by,

$$367 \quad S_l = \frac{1}{2} \sum_{i=1}^{l-1} \sum_{j=1}^{l-1} {}^1\beta_{i,j,l} Q_i^n Q_j^n - \sum_{i=1}^{l-1} {}^2\beta_{i,l} Q_i^n Q_l - \frac{1}{2} {}^3\beta_{l,l} Q_l^2 - \sum_{i=l+1}^{NS} {}^4\beta_{i,l} Q_i^o Q_l \quad (16)$$

368 where the superscript “n” denotes the “new” value which has been updated in the current loop.  
 369 By this method, the source term in the first equation of  $Q_1$  is calculated by the “old” values of  
 370 all the variables in the previous loop and then the value of  $Q_1$  is updated; the source term in the  
 371 second equation of  $Q_2$  is calculated by the updated  $Q_1$  and the “old” values of the other  
 372 variables in the previous loop and then the value of  $Q_2$  is updated. Following this line, the  
 373 source term in the  $l$ -th equation of  $Q_l$  is obtained by the updated variables from the sections

374 prior to the  $l$ -th section and the “old” values of the variables after the  $l$ -th section. In practice,  
375 after all the dependent variables  $Q_1 \cdots Q_{N_S}$  are obtained, one more solution procedure will be  
376 repeated as a correction process: the intermediate solution from the preceding step is  
377 employed instead of the solution from the last loop. Obviously, this two-stage hybrid-  
378 segregated scheme establishes a closer inter-equation coupling and is capable of avoiding the  
379 lagging effect considerably.

380 Another difficulty associated with the sectional population balance equation arises from the  
381 nonlinearity present in the source term. Inside each loop, the PDE is discretized in the  
382 computation domain and the resultant discretization equation is solved by iterative methods.  
383 However, the source term of the  $l$ -th section,  $S_l$ , is a nonlinear quadratic function of  $Q_l$  even  
384 though all the other variables are substituted by either “new” or “old” values in the previous or  
385 current loop. The nonlinear source term leads to troubles in constructing the discretization  
386 equation and also fails to employ the existing linear algebraic techniques. Thus the source term  
387 needs to be linearized before we proceed to construct and solve the discretization equation.

388 The simplest manner of linearizing the source term is to treat it as a constant irrespective of the  
389 iteration process, which is termed the explicit technique. This means that the source term is  
390 evaluated by the value of the dependent variable  $Q_l$  in the previous loop and keeps unchanged  
391 in the iteration process. This approach is acceptable only if the source is constant or relatively  
392 insignificant in comparison with other terms in the equation. However, for the ultra-fast  
393 coagulation/agglomeration processes such as acoustic agglomeration the source term is  
394 dominant in the equation, and the explicit technique has the issues of sluggish convergence and  
395 instability. In spite of unconditionally stability, on the other hand, the implicit technique is also  
396 not advisable having in mind the complexity of nonlinear solvers and computational efforts  
397 required. Thus, we can formally account for a linear dependence by decomposing the source  
398 term into the sum of an explicit part and an implicit part in the iterations which combines the  
399 merits of computational efficiency of explicit schemes and stability of implicit schemes. This  
400 technique is also termed as operator splitting method or fractional step method [14, 42, 45].

401 We denote the values of the previous iteration with a superscript “\*” to distinguish them from  
 402 the values of the previous loop. The source term at the current iteration can be linearized via a  
 403 first-order Taylor series,

$$404 \quad S_l = S_l(Q_l^*) + \underbrace{\left(\frac{\partial S_l}{\partial Q_l}\right)^*}_{\text{Implicit}} (Q_l - Q_l^*) = \underbrace{\left(\frac{\partial S_l}{\partial Q_l}\right)^*}_{\text{Implicit}} Q_l + \underbrace{S_l(Q_l^*) - \left(\frac{\partial S_l}{\partial Q_l}\right)^* Q_l^*}_{\text{Explicit}} \quad (17)$$

405 where, the first term is an implicit part and the second and third term are explicit. This  
 406 linearized form of the source term satisfies the method recommended by Patankar [41], which  
 407 can be written as

$$408 \quad S = S_C + \phi_P S_P \quad (18)$$

409 where,  $S$  is the source term  $S_l$ ,  $\phi_P$  is the dependent variable  $Q_l$  at the present computation node  
 410  $P$ ,  $S_P$  is the coefficient of  $\phi_P$  and  $S_C$  is a constant.

411 Substituting Eq. (16) into Eq. (17), we can immediately obtain,

$$412 \quad S_l = \underbrace{\left(-{}^3\beta_{l,l}Q_l^* - \sum_{i=1}^{l-1} {}^2\beta_{i,l}Q_i^n - \sum_{i=l+1}^{NS} {}^4\beta_{i,l}Q_i^o\right)}_{S_P} Q_l + \underbrace{\frac{1}{2} {}^3\beta_{l,l}(Q_l^*)^2 + \frac{1}{2} \sum_{i=1}^{l-1} \sum_{j=1}^{l-1} {}^1\beta_{i,j,l}Q_i^n Q_j^n}_{S_C} \quad (19)$$

413 where,  $S_C$  and  $S_P$  are expressed by

$$415 \quad S_C = \frac{1}{2} {}^3\beta_{l,l}(Q_l^*)^2 + \frac{1}{2} \sum_{i=1}^{l-1} \sum_{j=1}^{l-1} {}^1\beta_{i,j,l}Q_i^n Q_j^n \quad (20a)$$

$$416 \quad S_P = -{}^3\beta_{l,l}Q_l^* - \sum_{i=1}^{l-1} {}^2\beta_{i,l}Q_i^n - \sum_{i=l+1}^{NS} {}^4\beta_{i,l}Q_i^o \quad (20b)$$

417 However, updating the value of  $Q_l$  in every iteration indicates frequent access to CPU and the  
 418 memory. This will slow down the computation speed tremendously and cause enormous  
 419 burden of computation considering the large number of PDEs to be solved at all the  
 420 computation nodes. In practice, we use the value of  $Q_l$  in the previous loop as a constant to  
 421 evaluate the coefficients of linearization  $S_C$  and  $S_P$  until the next loop. In this way, the source  
 422 term is calculated by

$$S_l = \underbrace{\left(-{}^3\beta_{l,l}Q_l^o - \sum_{i=1}^{l-1} {}^2\beta_{i,l}Q_i^n - \sum_{i=l+1}^{NS} {}^4\beta_{i,l}Q_i^o\right)}_{S_P} Q_l + \underbrace{\frac{1}{2} {}^3\beta_{l,l}(Q_l^o)^2 + \frac{1}{2} \sum_{i=1}^{l-1} \sum_{j=1}^{l-1} {}^1\beta_{i,j,l}Q_i^n Q_j^n}_{S_C} \quad (21)$$

This linearization scheme is computationally efficient and stable with the splitting of implicit and explicit terms. It can also be seen that the implicit part of the source term is always negative-sloped ( $S_P < 0$ ), which increases the diagonal dominance of the resultant matrix of discretization equations and guarantees convergence and boundedness of the solution [46].

### 3.5 Solution of the sectional population balance equation

The sectional population balance equations are discretized by a finite volume based scheme and then solved by a linear algebraic solver in OpenFOAM. For transient problems, the temporal discretization is fulfilled by the Euler implicit method which guarantees boundedness and unconditional stability. When setting the boundary conditions, the wall can be regarded as perfectly absorbing with the value of  $Q_l$  equal to zero or impenetrable with the flux of  $Q_l$  equal to zero. The iterative solution procedure to solve the algebraic equations is the stabilized preconditioned bi-conjugate gradient (PBiCGStab), originally proposed by Van der Vorst [47]. In fact, the program in OpenFOAM supports the run-time selection (RTS) mechanism which enables the user to choose the discretization and solution schemes at a point of program execution [48]. The sectional population equation is solved in sequence from the first section to the last one with the decoupling of the source terms. Finally, the information regarding a general property of the aerosol is obtained for the complete set of particle size sections of interest.

The numerical methods of solving population balance equation are generally classified as the moment method, the Monte Carlo method and the discrete method. The moment method is superior in the low demands on the computational resources and relatively simple mathematical form for implementation. However, the inversion problem arises in the reconstruction of the distribution details from moment integrals [\*]. Plus, an additional limitation is usually imposed to the maximum time steps to address the realizability issue when the moment method is implemented in the Eulerian framework [\*]. The Monte Carlo method is

450 easy to incorporate multiple internal coordinates and can capture the history and trajectory of  
451 particles. However, the Monte Carlo method tracks the dynamics of each individual particle and  
452 thus compared to other methods, the computation is extremely expensive in computation time  
453 and memory usage [\*]. The sectional method, as one typical discrete method, is able to give the  
454 direct representation of the particle size distribution, and its computation cost is moderate,  
455 falling in between the moment method and the Monte Carlo method. Moreover, the solution  
456 process of the sectional method is conceptually straightforward in the finite-volume  
457 simulations and the “realizability” issue can be simply circumvented [\*]. However, since each  
458 size bin gives rise to one discrete population balance equation, the computation may be very  
459 expensive in time and memory when a large number of size bins are used.

#### 460 4. Description of the CFD-sectional algorithm

461 The solution procedure for the sectional PBE can be embedded into a typical CFD scheme such  
462 as the SIMPLE algorithm for steady-state problems or PISO algorithm for transient problems,  
463 which allows us to study the aerosol dynamics in a complex flow field. Additionally, the  
464 functionality of the CFD-sectional solver can be leveraged by incorporating with the thermal or  
465 turbulence models in OpenFOAM.

466 We shall describe the solution procedures of the CFD-sectional algorithm for the PBE coupled  
467 with incompressible flow equations (i.e., continuity equation and momentum equation) and  
468 focus on the solution of the sectional PBE. The important procedures can be summarized as  
469 follows:

- 470 1) Set up the initial conditions and boundary conditions for all the field variables.
- 471 2) Go through the SIMPLE loop for steady state flows or the PISO loop for transient flows:
  - 472 2a) Solution of pressure-velocity systems;
  - 473 2b) Solution of other transport equations such as turbulent quantities if present;
  - 474 2c) Solve the sectional coefficients via the modified numerical integration;
  - 475 2d) Calculate the sectional diffusivity (Brownian diffusivity and turbulent diffusivity if  
476 present);

477           2e) Calculate the coefficients of linearization  $S_C$  and  $S_P$  using the “new” values for the  
478 lower sections and the “old” values for the current or higher sections;

479           2f) Linearize the source term by splitting it into the sum of implicit and explicit parts;

480           2g) Solve the sectional population balance equation via a linear algebraic iteration  
481 technique;

482           2h) Proceed to the next section until the sectional PBE is solved for all sections;

483           2i) Repeat the solution procedure from 2d) to 2h) to obtain the corrected solutions.

484   3) If the convergence criterion or the final time step is not reached, return to step 2).

485   It is to be noted that the step 2c) can be executed out of the SIMPLE or PISO loop if the  
486 sectional coefficients are independent on the flow characteristics or fluid properties. The  
487 flowchart of the developed CFD-sectional solver is illustrated in Fig. 4. The solution of flow  
488 equations for the velocity-pressure system is represented by  $U$  Equation and  $p$  Equation, and  
489 the solution process of the sectional population balance equation is represented by  $Q$  Equation.  
490 The standard  $k$ - $\varepsilon$  model, which introduces two extra transport equations for turbulent kinetic  
491 energy,  $k$  Equation and turbulent dissipation rate,  $\varepsilon$  Equation, is selected for Reynolds-averaged  
492 simulation (RAS) turbulence modeling.

493

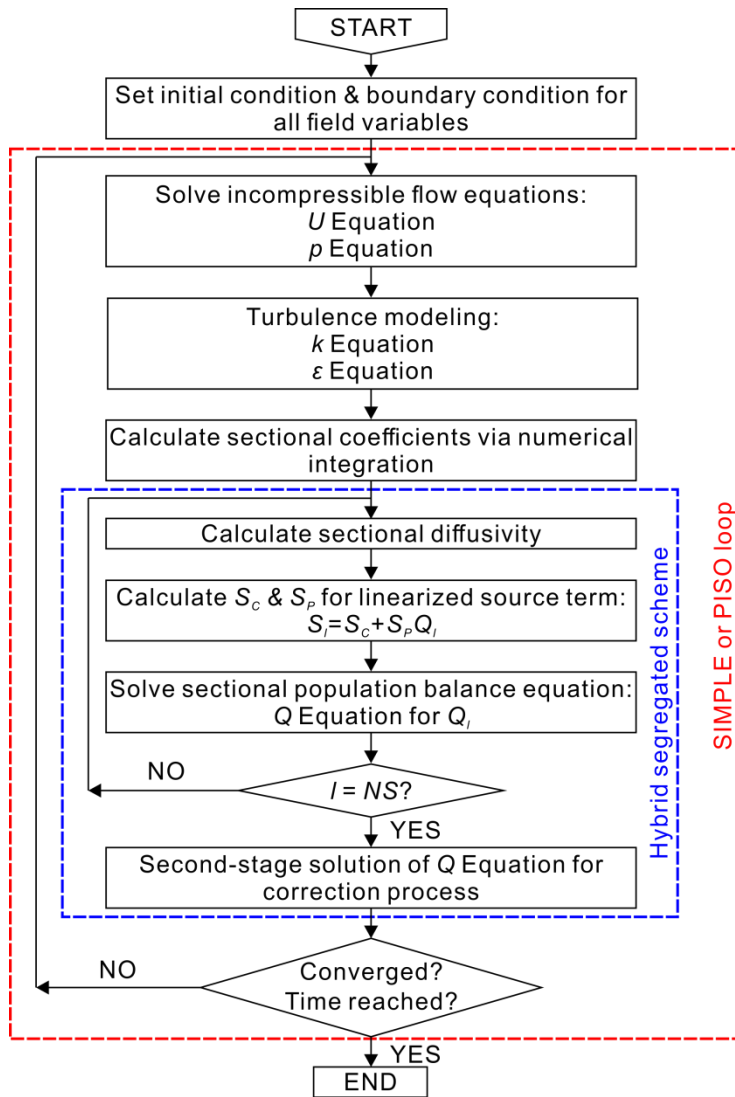


Fig. 4. Schematic representation of the CFD-sectional algorithm.

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## 5. Results and Discussion

499

### 5.1. Validation of the CFD-sectional algorithm

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#### 5.1.1. The modified numerical integration

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To improve the computation efficiency and accuracy, the sectional coefficients are solved by

502

some numerical quadrature which is modified by adaptively adjusting the integration limits

503

corresponding to the discontinuous functions. In this section, we will examine the improvement

504

of the modified numerical integration by considering a simple situation where  $Q_i$  represents the

505

particle number concentration  $N_i$ , and the kernel is equal to a constant. Under such situation,

506 the sectional coefficients in Table 1 can be directly figured out with exact values. Without loss  
 507 of generality, the kernel is regarded as equal to unity, and then the sectional coefficients are  
 508 equal to the ratio of the area of the “goal” region determined by the discontinuous functions to  
 509 the area of the original rectangle as shown in Fig. 1-3. The detailed expressions for the  
 510 numerical integrals and exact values of the sectional coefficients when  $Q_l = N_l$  and  $\beta = 1$  are  
 511 listed in the Supplementary Material.

512 Let us take an exemplary size domain of  $\zeta = \log_{10}[d_p(\mu m)] = (-1.05, 1.05)$  divided by 20  
 513 sections. The sectional coefficients are solved by the composite Simpson quadrature for the  
 514 same number of integration nodes with and without the adaptive integration limits.  ${}^1\beta_{i,j,20}$ ,  
 515 the “birth” rate of the particles in the 20-th section, is investigated as an example. For such a  
 516 geometric sectioning scheme,  ${}^1\beta_{i,j,20}$  is equal to zero when  $i < 19, j < 19$  because at least one  
 517 of a pair of colliding particles is required to come from the 19-th section to generate a resultant  
 518 particle in the 20-th section as aforementioned. Thus, the selected results for  $i = 19, j =$   
 519  $1, 3 \dots 19$  can be presented in Table 3 as the representative values of  ${}^1\beta_{i,j,20}$  since  ${}^1\beta_{i,j,20} =$   
 520  ${}^1\beta_{j,i,20}$ . It can be seen that the results of the numerical integration without adaptive limits are  
 521 basically coincident with the exact values for  $j > 13$  but deviate seriously from the exact values  
 522 for  $j \leq 13$ . Obviously, the deviation from exact values is increased when the  $j$ -th section goes  
 523 far away from the 19-th section. Especially, when  $j$  is smaller than 11, there will be only one  
 524 integration node lying inside the “goal” region, leading to a constant with a serious error of  
 525 computation. With the adjusted integration limits, the modified numerical integration gives rise  
 526 to satisfactory results and the relative error is much smaller than 1% for all pairs of colliding  
 527 sections, which indicates that the accuracy of the calculation is greatly improved and the  
 528 adaptive adjustment of the integration limits effectively relieves the computation error caused  
 529 by the skewness of colliding sections.

530 Table 3. Selected  ${}^1\beta_{19,j,20}$  calculated by numerical integration with and without adaptive limits.

$j$	$\zeta$	$v$ ( $\mu m^3$ )	Exact	Beta1-W/ Error-W/ (%)	Beta1-W/O Error-W/O (%)
1	-0.945	7.656E-22	3.076E-06	3.076E-06 0.003	3.333E-03 108274.865
3	-0.735	3.266E-21	1.312E-05	1.312E-05 0.003	3.333E-03 25305.548
5	-0.525	1.393E-20	5.597E-05	5.597E-05 0.003	3.333E-03 5855.642

7	-0.315	5.943E-20	2.388E-04	2.388E-04	0.004	3.333E-03	1296.139
9	-0.105	2.535E-19	1.018E-03	1.019E-03	0.004	3.333E-03	227.287
11	0.105	1.081E-18	4.345E-03	4.345E-03	0.003	3.333E-03	23.276
13	0.315	4.613E-18	1.853E-02	1.853E-02	0.003	1.922E-02	3.718
15	0.525	1.968E-17	7.906E-02	7.906E-02	0.003	7.918E-02	0.151
17	0.735	8.395E-17	3.372E-01	3.373E-01	0.003	3.372E-01	0.008
19	0.945	3.581E-16	9.981E-01	9.979E-01	0.026	9.979E-01	0.026

531

### 532 5.1.2. Validation against analytical solution of Brownian coagulation/agglomeration

533 We validate the present CFD-sectional algorithm against an analytical solution of the self-  
534 preserving distribution of an aerosol undergoing Brownian coagulation/agglomeration. Lee et al.  
535 [49] studied the change of particle size distribution caused by Brownian  
536 coagulation/agglomeration in the continuum regime, and obtained a self-preserving analytical  
537 solution for the size distribution of an initially lognormal distribution.

538 The self-preserving particle size distribution in terms of the logarithmic coordinates of the  
539 particle diameter is described by

$$540 \quad n_p(\zeta, t) = \frac{dN}{dv} \frac{dv}{d\zeta} = \frac{\ln(10)N(t)}{\sqrt{2\pi}\ln\sigma(t)} \exp\left\{-\frac{\ln^2[d_p/d_{pg}]}{2\ln^2\sigma(t)}\right\} \quad (22)$$

541 where,  $\zeta$  is the size variable,  $\zeta = \log_{10}d_p = \log_{10}[(6v/\pi)^{1/3}]$ . At the time  $t$ , the total number  
542 concentration  $N(t)$ , the geometric mean particle diameter  $d_{pg}$ , and the geometric standard  
543 deviation  $\sigma(t)$  are described by [49],

$$544 \quad \frac{N(t)}{N_0} = \frac{1}{1+[1+\exp(\ln^2\sigma_0)]KN_0t} \quad (23a)$$

$$545 \quad \left(\frac{d_{pg}}{d_{pg0}}\right)^3 = \frac{\exp\left(\frac{9}{2}\ln^2\sigma_0\right)\{1+[1+\exp(\ln^2\sigma_0)]KN_0t\}}{\langle 2+\{\exp(9\ln^2\sigma_0)-2\}/\{1+[1+\exp(\ln^2\sigma_0)]KN_0t\}\rangle^{1/2}} \quad (23b)$$

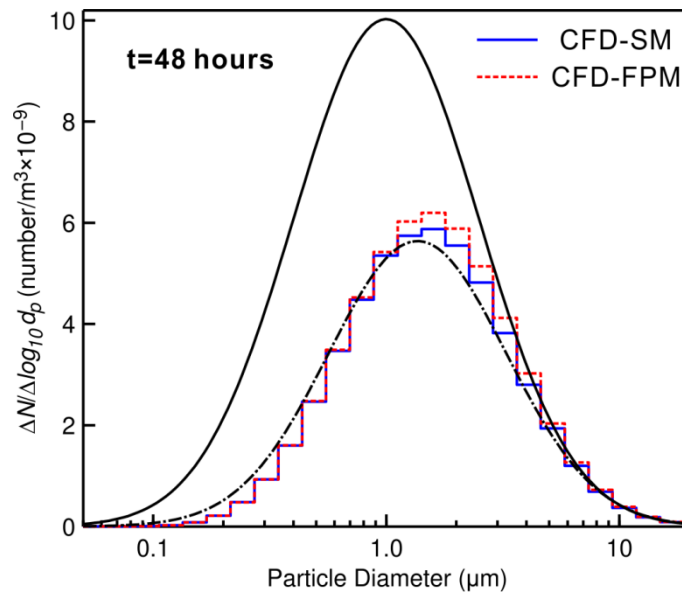
$$546 \quad \ln^2\sigma(t) = \frac{1}{9} \ln \left[ 2 + \frac{\exp(9\ln^2\sigma_0)-2}{1+[1+\exp(\ln^2\sigma_0)]KN_0t} \right] \quad (23c)$$

547 where  $K$  is a constant,  $K = 2k_B T/3\mu$ , and  $N_0$ ,  $d_{pg0}$  and  $\sigma_0$  are the initial values for  $N(t)$ ,  $d_{pg}$   
548 and  $\sigma(t)$ , respectively.

549 The evolution of an aerosol with particles undergoing Brownian coagulation/agglomeration in  
 550 the continuum regime is simulated by the CFD-sectional solver with the kernel described by,

$$551 \quad \beta(u, v) = \frac{2k_B T}{3\mu} [(u)^{1/3} + (v)^{1/3}] \left[ \frac{1}{(u)^{1/3}} + \frac{1}{(v)^{1/3}} \right] \quad (24)$$

552 where the Cunningham coefficient is equal to unity in the continuum regime. For comparison  
 553 with the analytical solution, the flow velocity is set to zero and the aerosol is homogeneous in  
 554 space. The operation temperature for Brownian coagulation is 293 K. The aerosol at the initial  
 555 time is log-normal with a geometric mean diameter of 1  $\mu\text{m}$  and standard deviation of 2.5. The  
 556 number concentration  $N_0$  at the inlet is  $10^{10}$  particles/ $\text{m}^3$  and the particle density is 1,000  
 557  $\text{kg}/\text{m}^3$ . The simulation was performed in parallel on a workstation equipped with 8 CPUs (Intel  
 558 i7) and the tolerance is  $1\text{e-}6$  for all variables. The CPU time is 14,058 seconds for the  
 559 computation domain of 10,000 grids to complete 48 hours of physical time. The simulation  
 560 results were compared against those of the CFD-fixed pivot model in terms of the computation  
 561 time, accuracy and algorithm complexity. To be in line with the analytical solution, the number  
 562 density function for the CFD-fixed pivot method is calculated by  $\bar{n}_i = N_i / (v_i - v_{i-1})$  for the  $i$ -  
 563 th size bin, where  $N_i$  is the number concentration, and  $v_{i-1}$  and  $v_i$  are size bin boundaries.



564  
 565  
 566 Fig. 5. Comparison between the self-preserving particle distribution (dash-dotted line) and the  
 567 simulation results predicted by the CFD-sectional method (solid stairs) and the CFD-fixed pivot method  
 568 (dashed stairs) for an aerosol of particles undergoing Brownian coagulation in the continuum regime at  
 569 48 hours. The initial distribution is denoted by the solid line.

570  
571 The particle size distributions predicted by the present CFD-sectional method (solid stairs) and  
572 the CFD-fixed pivot method (dashed stairs) at 48 hours are compared with the analytical  
573 solution (dash-dotted line) and the initial distribution (solid line) in Fig. 5. It can be observed  
574 that the particle size distributions predicted by both methods match the analytical solution of  
575 the self-preserving distribution very well, and the discrepancy between simulations and exact  
576 solutions is expected to become smaller when wider particle size range is considered. The  
577 satisfactory agreement verifies the accuracy of the present CFD-sectional approach and  
578 computation codes in the solution of the population balance equation. It is also noticeable that  
579 the CFD-sectional simulation shows better agreement with the self-preserving solution than the  
580 CFD-fixed pivot method. As shown in Table 4, the geometric mean diameter and geometric  
581 standard deviation are 1.489  $\mu\text{m}$  and 2.274 for the CFD-sectional simulation, and 1.527  $\mu\text{m}$  and  
582 2.267 for the CFD-fixed pivot method, compared to 1.362  $\mu\text{m}$  and 2.406 for the analytical self-  
583 preserving solution, respectively. Obviously, the CFD-sectional method has higher accuracy than  
584 the CFD-fixed pivot method with the same number of size bins (sections or pivots).

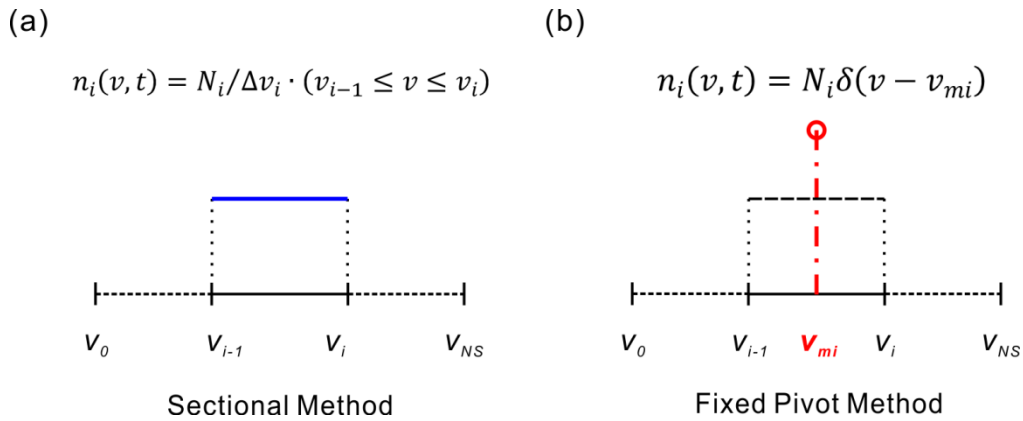
585 Table 4. Comparison of the numerical results predicted by the CFD-sectional method and CFD-fixed pivot  
586 method.

Model	Geometric Mean Diameter		Geometric Standard Deviation		CPU Time (second)
	$d_g$ ( $\mu\text{m}$ )	Error (%)	$\sigma$	Error (%)	
Analytical	1.362	N.A.	2.406	N.A.	N.A.
CFD-SM	1.489	9.32%	2.274	5.49%	14,058
CFD-FPM	1.527	12.11%	2.267	5.78%	9,430

587  
588 Both the sectional method and the fixed pivot method are based on the discrete method while  
589 the major difference lies in the pre-defined functional form within each size bin. As shown in  
590 Fig. 5a, the sectional method assumes that the particle size distribution is uniformly distributed  
591 within each size bin, indicating that the assumed distribution function in the  $i$ -th size bin is given  
592 by,  $n_i(v, t) = N_i/\Delta v_i \cdot (v_{i-1} \leq v \leq v_i)$ , where,  $N_i$  is the number concentration,  $\Delta v_i = v_i -$   
593  $v_{i-1}$  is the width of the  $i$ -th size bin. The fixed pivot method, as shown in Fig. 5b, assumes that  
594 the particle distribution within each size bin is concentrated at a representative size or pivot in  
595 the spirit of point mass, which means that the distribution function is a singular point function,  
596  $n_i(v, t) = N_i\delta(v - v_{mi})$ , where,  $v_{mi}$  is the pivot and  $\delta$  is the Dirac delta function, with  $\delta = \infty$

597 when  $v = v_{mi}$  and otherwise  $\delta = 0$ . Theoretically, the uniform distribution assumed by the  
 598 sectional method provides higher-order accuracy than the singular point distribution at pivots  
 599 in the fixed pivot technique. Thus, the CFD-sectional simulation shows better agreement with  
 600 the self-preserving solutions than the CFD-fixed pivot method. However, due to the complex  
 601 functional form of the particle distribution, the solution of the sectional coefficients involves  
 602 troublesome numerical quadrature of double integrals with discontinuous integrands for the  
 603 sectional method. In contrast, the reassignment coefficients (equivalent to sectional  
 604 coefficients) of the fixed pivot method can be easily solved via simple algebraic operations.  
 605 Thus, in spite of the modified numerical quadrature with adaptive integration limits, the CFD-  
 606 sectional method demands more computation time than the CFD-fixed pivot method as shown  
 607 in Table 4.

608



609

610

Fig. 6. Illustration of the sectional method (a) and the fixed pivot method (b).

611

612 The zeroth, second, third and sixth moments of a particle size distribution are proportional to  
 613 the total number, total surface area, total volume and total volume square of particles per unit  
 614 space, respectively. The present CFD-sectional solver conserves the total volume/mass, and the  
 615 other properties can be derived readily based on the  $v$ -model [38, 39]. It can be seen in Table 5  
 616 that the approximate values of the derived properties based on the volume-conserved model  
 617 are in accordance with the analytical solution for both low-order and high-order properties, and  
 618 the errors are 0.200%, 1.923% and 1.997%, for the total number,  $N$ , total surface area,  $S$  and  
 619 total volume square,  $v^2$ , respectively. Thus, the present solver is shown to have excellent  
 620

621 performance in terms of the three common derived properties or moments. Generally, the  
 622 accuracy of the sectional method is determined by a combination of many factors, such as the  
 623 sectional coefficients, the discretization scheme, the internal consistency and sectioning space.  
 624 The present study justifies that for the derived moments or properties, the deviation caused by  
 625 the internal inconsistency is not significant if the sectioning space is fine enough [39]. Vanni [40]  
 626 made similar conclusion via the comparison of the number averaged particle size and the  
 627 volume/mass averaged particle size.

628 Table 5. Comparison of total number, total surface area and total volume square per unit space between  
 629 the analytical solution and the derived values based on the CFD-sectional solver.

	$N$ (particles/unit space)	$S$ (m <sup>2</sup> /unit space)	$v^2$ (m <sup>6</sup> /unit space)
CFD-sectional	6.992E9	0.159	5.515E-21
Analytical Solution	7.006E9	0.156	5.407E-21
Error	0.200%	1.923%	1.997%

630

## 631 5.2. Case study of acoustic agglomeration

632 Acoustic agglomeration, in recent years, has gained world-wide attention, as an efficient  
 633 preconditioning technique in air conditioning systems [17, 50-53]. Under the action of a sound  
 634 field, small particles in a suspension collide with one another to form large agglomerates and as  
 635 a result, the efficiency of conventional particle removal technologies is greatly improved. The  
 636 particle size distribution can shift significantly to larger particulates within a short time of  
 637 several seconds in the acoustic chamber, which indicates that the acoustic agglomeration is an  
 638 ultra-fast coagulation/agglomeration process. In this section, a numerical analysis of acoustic  
 639 agglomeration in an inhomogeneous velocity field is carried out by the CFD-sectional algorithm  
 640 and the numerical simulation is compared with the experimental results reported by Yan et al.  
 641 [52] regarding acoustic agglomeration in a standing wave field.

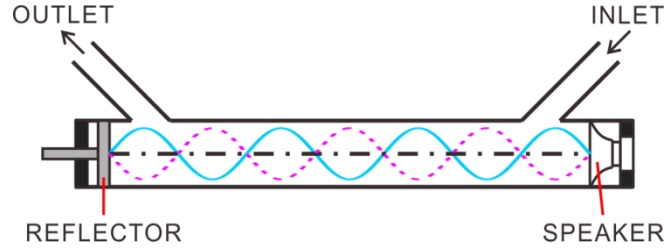
### 642 5.2.1. Acoustic kernels

643 The acoustic treatment in a standing-wave agglomerator can achieve higher particle removal  
 644 efficiency than a travelling field under the same power consumption. As shown in Fig. 7, the  
 645 acoustic velocity of the air at position  $x$  and time  $t$  of a plane standing wave field by the  
 646 superposition of an incident wave and reflected wave can be described as

$$647 \quad u_g(x, t) = U_{g0} \sin(kx) \sin(2\pi ft) \quad (25)$$

648 where,  $U_{g0}$  is the acoustic velocity amplitude,  $k$  is the wave number and  $f$  is the sound  
 649 frequency.

650



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653 Fig. 7. Illustration of a standing wave field in an acoustic agglomerator.

654

655 A variety of mechanisms have been proposed for the interaction between the sound field and  
 656 suspended particles in aerosols and two primary mechanisms are generally accepted, which are  
 657 the so-called orthokinetic and hydrodynamic interactions [17]. The acoustic kernels for two  
 658 particles of diameter  $d_1$  and  $d_2$  which undergo orthokinetic and hydrodynamic interactions in a  
 659 standing wave field can be calculated by

660

$$\beta_S^{OI}(d_1, d_2, x) = 1/2 (d_1 + d_2)^2 U_{g0} \eta_{12} |\sin(kx)| \quad (26a)$$

661

$$\beta_S^{HI}(d_1, d_2, x) = \frac{\sqrt{3} \rho_g U_{g0}^2 \sin^2(kx)}{144 \pi \mu} \frac{d_1^2 d_2^2}{d_1 + d_2} \quad (26b)$$

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where,  $\rho_g$  is the air density,  $\mu$  the dynamic viscosity of the air, and  $\eta_{12}$  is the entrainment factor associated with the sound frequency and sound intensity, which are given in Shang et al. [17]. It can be seen that the acoustic kernels in a standing wave field are varied along the propagation direction and accordingly the acoustic agglomeration becomes nonhomogeneous in space. Additionally, the acoustic kernels for orthokinetic and hydrodynamic mechanism are of the order of  $10^{-4}$  cm<sup>3</sup>/s or higher, which are much larger than the Brownian kernel ( $10^{-9} - 10^{-8}$  cm<sup>3</sup>/s). Therefore, the Brownian kernel is negligible in acoustic agglomeration and only the acoustic kernel – the additive combination of Eq. (26a) for orthokinetic interaction and Eq. (26b) for hydrodynamic interaction is considered in the present study. In the present study, the particles are relatively large (0.1 – 10  $\mu$ m) compared to the mean free path (0.066  $\mu$ m) of air molecules, and thus, the Cunningham slip correction has insignificant impact on the acoustic

673 kernels and can be neglected. When particles are even smaller to be comparable with the mean  
674 free path of air molecules, the slip correction must be taken into account [10].

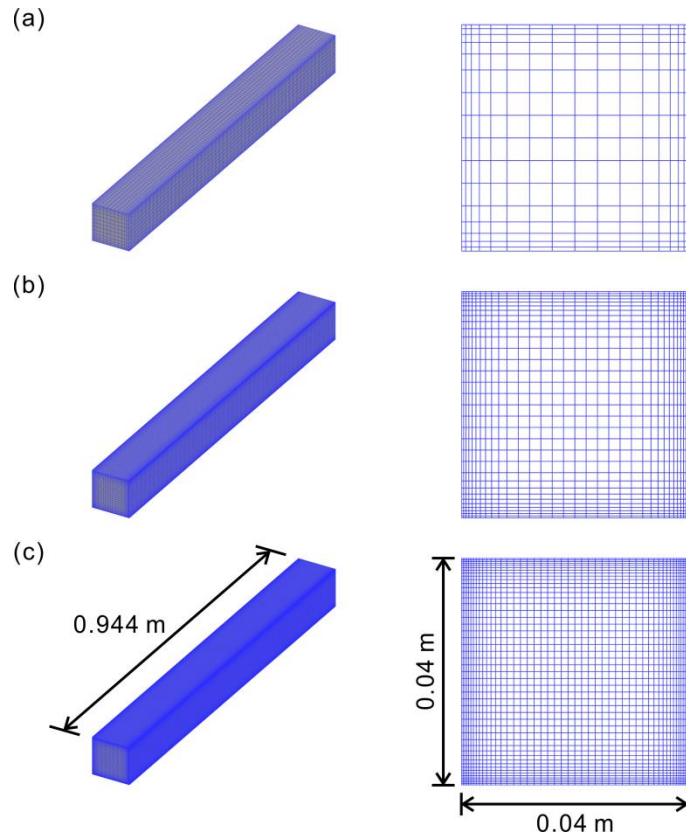
### 675 5.2.2. Governing equations, meshing scheme and numerical settings

676 In acoustic agglomeration, due to the periodic acoustic wave, both the flow characteristics  
677 (transport velocity) and aerosol properties (PSD and concentration) fluctuate periodically with  
678 time around their time-average values. Thus, the physical variables in the governing equations  
679 consist of two components: the time-averaged value and the fluctuation induced by the  
680 acoustic wave within each period of acoustic cycle. Thus, to fully resolve the flow characteristics  
681 and aerosol properties over an acoustic period, the flow field induced by the acoustic wave has  
682 to be solved within each period of acoustic cycle. However, under such situation, the  
683 computation will be too expensive to afford for the current computer capacity when we  
684 consider the large number of PDEs, time steps and meshes. Plus, the sampling time of typical  
685 aerosol measurement instruments is in the order of tens of seconds, which results in that the  
686 fluctuations of the particle size distribution within one acoustic period will be filtered out in  
687 measurements. Thus, even if we obtained the flow field and particle size distribution which are  
688 time-resolved within one acoustic period, the numerical predictions cannot be validated by the  
689 experimental measurements. Therefore, in the present study, we overlook the fluctuations of  
690 flow velocity within one acoustic period and solve the time-averaged flow equations without  
691 the flow field induced by the acoustic wave. Similarly, the time-averaged population balance  
692 equation is solved without considering the fluctuations of PSD within one acoustic period.  
693 Similar to the Reynolds averaging in turbulent flows, we only average the periodic fluctuations  
694 of physical variables over one acoustic period and the time-dependence due to the operating  
695 conditions, such as the inflow, the sound frequency or SPL etc., can be resolved. Since the  
696 nonlinear high-order acoustic effects will be significant only for ultrasound frequencies at tens  
697 of kilo Hertz [10], we can neglect the high-order term resulting from the averaging process for  
698 this case study (1.8 kHz for the sound frequency) without introducing significant errors. As such,  
699 the time-averaged governing equations for the flow velocity and population balances share the  
700 same form as the expression in Eq. (5), as demonstrated in the Supplemental Material. Hence  
701 the developed CFD-sectional solver can be used directly except that the initial and boundary

702 conditions should be set for the time-averaged flow and particle size distribution. In Yan et al.  
703 [52], the operating conditions such as the inflow, sound frequency, SPL and PSD etc. at the inlet  
704 do not vary with time and thus the agglomeration process after averaging over one acoustic  
705 cycle can be regarded as steady state.

706 In the typical case of Yan et al. [52], the acoustic agglomerator is a duct of 0.04 m width and  
707 0.04 m height. The distance between the inlet and sampling point is equal to 5 wavelengths  
708 (0.944 m) and the residence time is 2.5 s. The dimensions of the computation domain for the  
709 numerical simulation are identical to those of the agglomerator in the experiments. To test the  
710 mesh independence, we study three levels of mesh resolution based on the same meshing  
711 scheme as illustrated in Fig. 8: coarse (16 length  $\times$  16 width  $\times$  180 height), standard (32 length  
712  $\times$  32 width  $\times$  360 height), and fine (48 length  $\times$  48 width  $\times$  540 height). Meshes near the walls  
713 are elaborately refined to capture the sharp gradients. The aerosol is re-dispersed coal ash with  
714 the diameter of the particles between 0.023  $\mu\text{m}$  and 9.314  $\mu\text{m}$ . The sound frequency of the  
715 standing wave field is 1.8 kHz and the sound pressure level is 150 dB at every half wavelength  
716 from the acoustic source. The operation pressure and temperature are 1 atm and 293 K,  
717 respectively. The simulation was performed in parallel on a workstation equipped with 8 CPUs  
718 (Intel i7) and the tolerance is  $1\text{e-}6$  for all variables.

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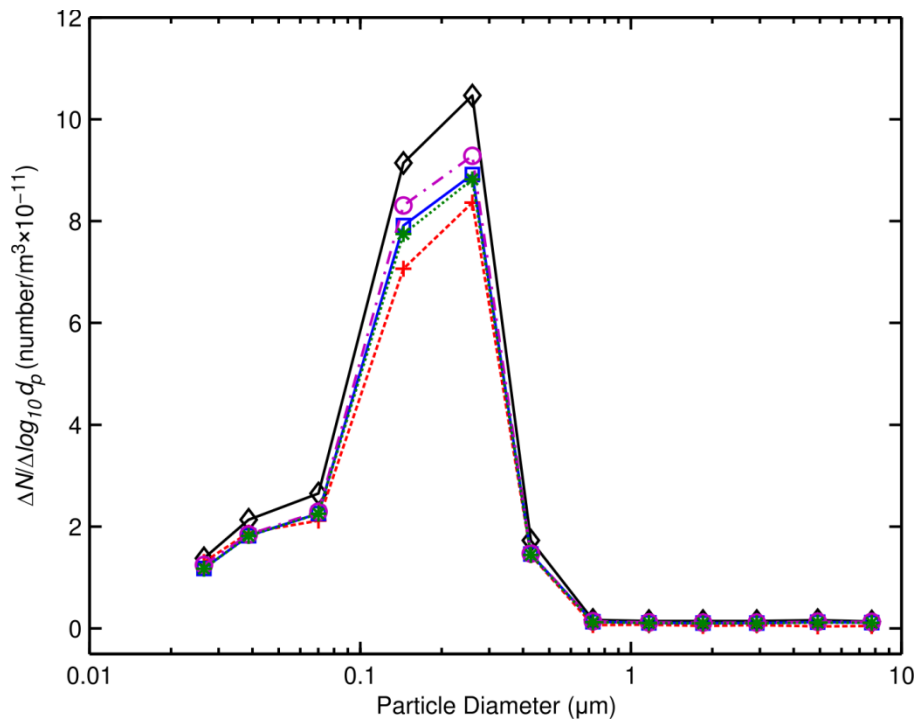
Fig. 8. Illustration of the computation domain and meshing scheme in the mesh independence tests (not to scale). (a) coarse (16 width  $\times$  16 depth  $\times$  180 length); (b) standard (32 width  $\times$  32 depth  $\times$  360 length); (c) fine (48 width  $\times$  48 depth  $\times$  540 length).

726 Initially, no particles exist inside the acoustic agglomerator. The air velocity at the inlet is  
727 calculated from the agglomerator length and residence time of particles. The particle size  
728 distribution at the inlet is the same as the initial measurement of Yan et al. [52]. At the outlet,  
729 the flow is assumed to be fully developed and all other scalar variables ( $Q_s$ ) are set as the zero  
730 Gradient (Neumann) boundary condition. The air on the walls is non-slip and impenetrable. The  
731 particle concentration is set as zero on the walls (Dirichlet boundary condition), which implies  
732 that the surface is a perfect sink for deposition, without resuspension or reentrainment of  
733 particles. The particle concentration is set as zero on the walls (Dirichlet boundary condition),  
734 which implies that the surface is a perfect sink for deposition, without resuspension or  
735 reentrainment of particles.

736 5.2.3. PSD, number concentration and agglomeration efficiency

737 The average particle size distribution at the outlet of the acoustic chamber is demonstrated in  
738 Fig. 9 for the numerical simulations with the coarse (circle), standard (square) and fine (star)  
739 meshes in comparison to the experimental measurement (plus). The numerical simulation was  
740 conducted under the same initial distribution (diamond) as in the experiments. It can be seen  
741 that the simulation results with three levels of meshing resolution reveal a similar trend that  
742 the number concentration of the particles in submicron range decreases strikingly, indicating  
743 the occurrence of the particle agglomeration. However, there is a noteworthy deviation  
744 between the prediction with coarse meshes and the simulation results with the standard or fine  
745 meshes. A subtle difference is detected from the standard to fine meshes, confirming that the  
746 solution with the standard resolution can be considered “mesh independent”. Even though the  
747 accuracy is slightly improved for the fine meshes, the computation time is increased  
748 dramatically from 2,471 seconds for standard meshes to 19,278 seconds for fine meshes.  
749 Therefore, the simulation with standard meshes provides a good trade-off between the  
750 accuracy and computational cost, and is used in the following study.

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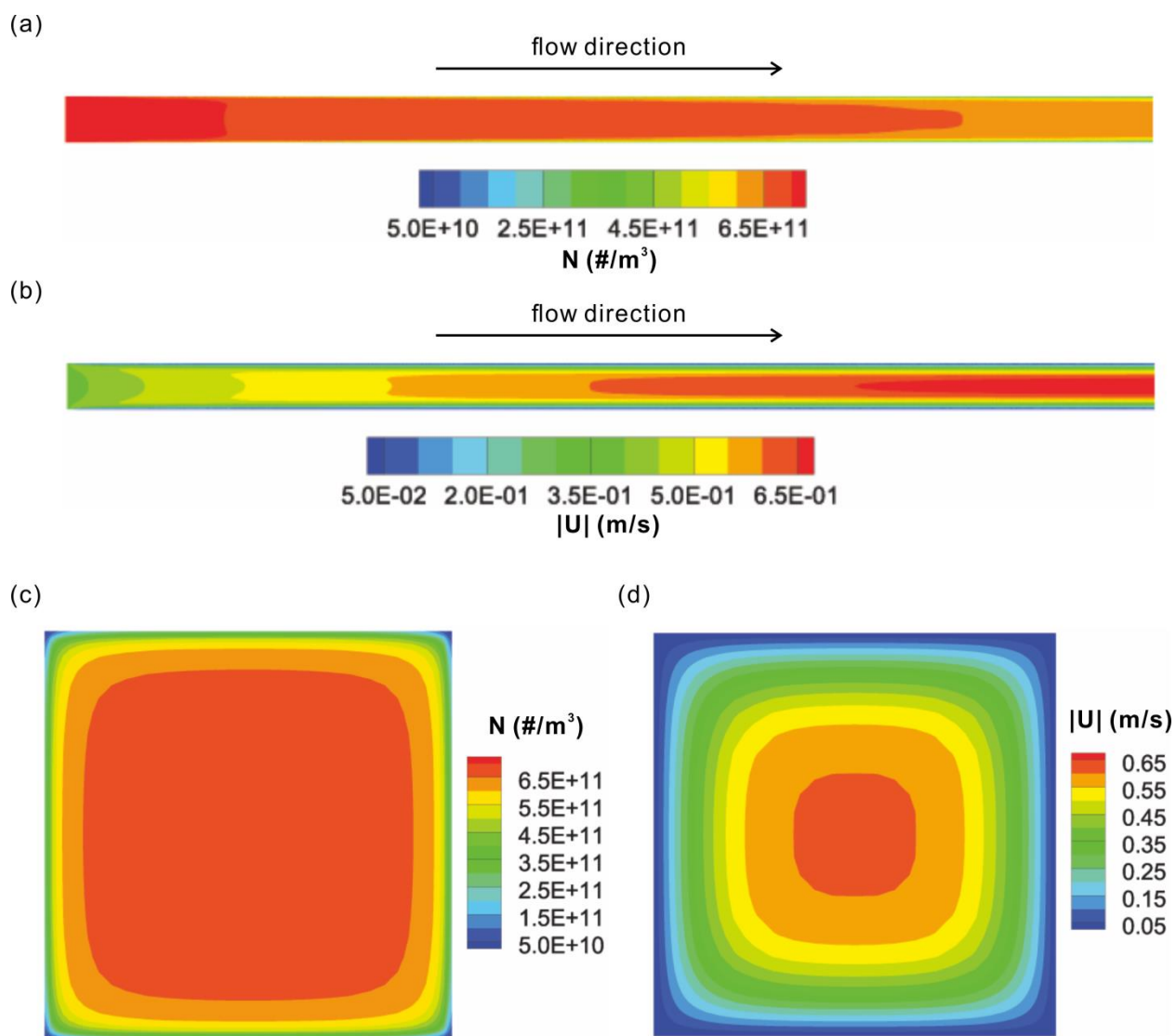
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754 Fig. 9. Simulation results compared with experimental measurements (plus) of the particle size  
755 distribution in the acoustic agglomeration for coarse (circle), standard (square) and fine (star) meshes.  
756 The initial distribution (diamond) of the aerosol at the chamber inlet is also included.  
757

758 In Fig. 9, the particle size distribution predicted by the standard or fine meshes shows a  
759 satisfactory agreement with the experimental data except that a relatively large deviation is  
760 observed for the range from 0.1 to 0.4  $\mu\text{m}$ . In fact, the accuracy of the present CFD-sectional  
761 method is also associated with the number of discretization sections for the population balance  
762 modeling. As shown in Fig. 9, the particle size distribution in the range of 0.1 – 0.4  $\mu\text{m}$  is varied  
763 rapidly with a much steeper slope than that out of this size range. Thus, to ensure the same  
764 level of accuracy, the particle size in this range should be discretized with more refined  
765 sectioning space, i.e., a larger number of sections are required. However, restricted by the  
766 measurement data in experiments [52], we cannot divide the particle size in this range into very  
767 fine sections in the simulation. Therefore, the discrepancy between the numerical and  
768 experimental results in the range of 0.1 – 0.4  $\mu\text{m}$  is greater than that out of this size range. The  
769 deviation in this range is expected to be smaller than what it is currently when finer sectioning  
770 spaces are adopted for the population balance modeling. In addition, the air-particle and  
771 particle-particle mechanisms for acoustic agglomeration have not been well understood, which  
772 also limits the predictive accuracy of the current acoustic kernels in covering a wide range of  
773 experimental conditions [\*]. Some experimental factors such as the losses of particles in  
774 sampling hoses of the measuring devices cannot be incorporated in the simulation, which may  
775 be another source of the deviation.

776 The total number concentration of the particles and the flow velocity are presented in Fig. 10  
777 for the contours along and across the flow direction. It can be seen from Figs. 10a and 10c that  
778 the number concentration drops down evidently along the flow direction as the small particles  
779 collide and form into agglomerates with the progress of acoustic agglomeration. In the cross  
780 section of the duct the number concentration is distributed non-uniformly – equal to its  
781 minimum at the wall and to maximum at the core region, and the contours of the number  
782 concentration are similar to the velocity field in Figs. 10b and 10d. But the contours are not  
783 completely in accordance with each other: From the wall to the core region, the number  
784 concentration of the particles undergoes a much steeper increase than the flow velocity.

785 Although the acoustic kernel is dependent on the propagation distance only, the  
 786 inhomogeneous velocity and boundary deposition will result in the non-uniformity of the  
 787 particle distribution across the flow direction. Hence, the complex distribution of the particles is  
 788 determined by not a single factor, but the combined effects of the velocity field (convection  
 789 term), wall deposition (boundary condition) and acoustic agglomeration (source term).



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 792 Fig. 10. (a) Spatial distribution of the total number concentration of the particles along the flow  
 793 direction. (b) Spatial distribution of the velocity field along the flow direction. (c) Spatial distribution of  
 794 the total number concentration of the particles across the flow direction. (d) Spatial distribution of the  
 795 velocity field across the flow direction.

796  
 797 For quantitative analysis, the efficiency for acoustic agglomeration can be defined by

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$$\eta = (1 - N/N_0) \times 100\% \quad (27)$$

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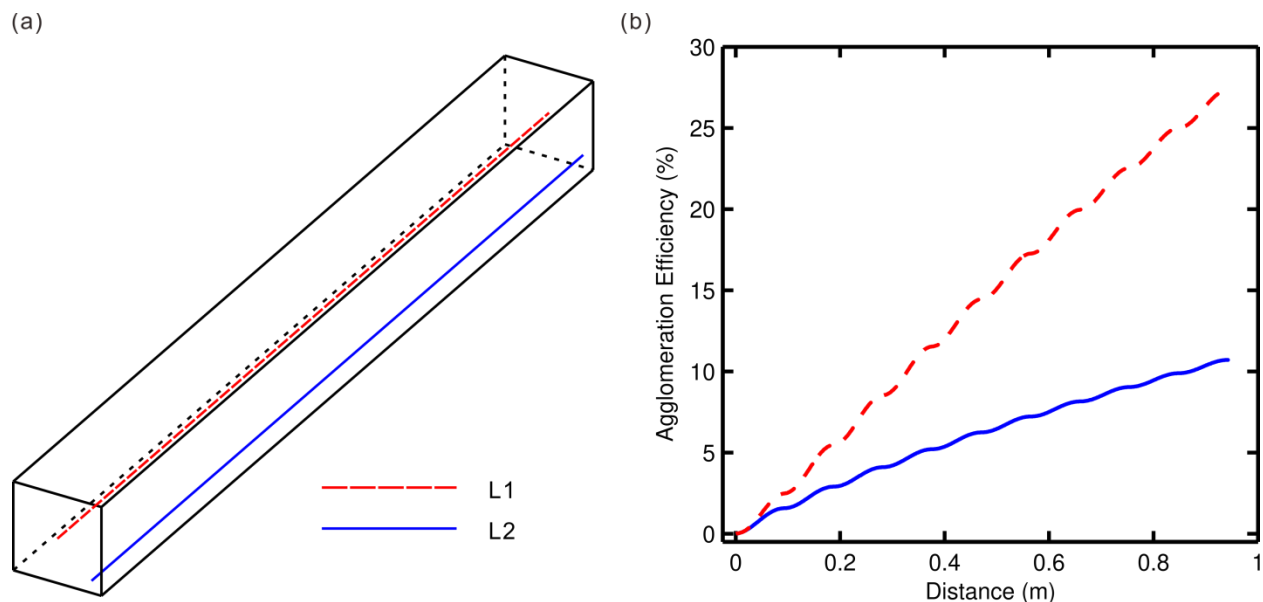
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where  $N$  and  $N_0$  are the total number concentration of the particles at some position and at the inlet, respectively. The average agglomeration efficiency in the cross section is 18.2%, which is agreeable with the one-dimensional simplified model [17]. To demonstrate its cross-sectional distribution at different positions, the agglomeration efficiency is plotted along two straight lines in the flow direction as shown in Fig. 11a: the center line, L1 (solid), with  $y=0.02\text{m}$  and  $z=0.02\text{m}$  and a representative line near the wall, L2 (dashed), with  $y=0.002\text{m}$  and  $z=0.002\text{m}$ . As observed in Fig. 11b, the agglomeration efficiency of the particles increases from 0 at the inlet to 27.4 % for L1 and to 10.7% for L2 at the outlet. In closer proximity to the wall, the flow velocity is slower and then the particles experience a longer time of acoustic agglomeration. Thus, the acoustic agglomeration efficiency along L1 is larger than that along L2. It is also worth noting that the agglomeration efficiency is not increased linearly but displays node-antinode alternations in an intermittent way, which is ascribed to the spatial alternation of the acoustic kernel from the velocity node to antinode.



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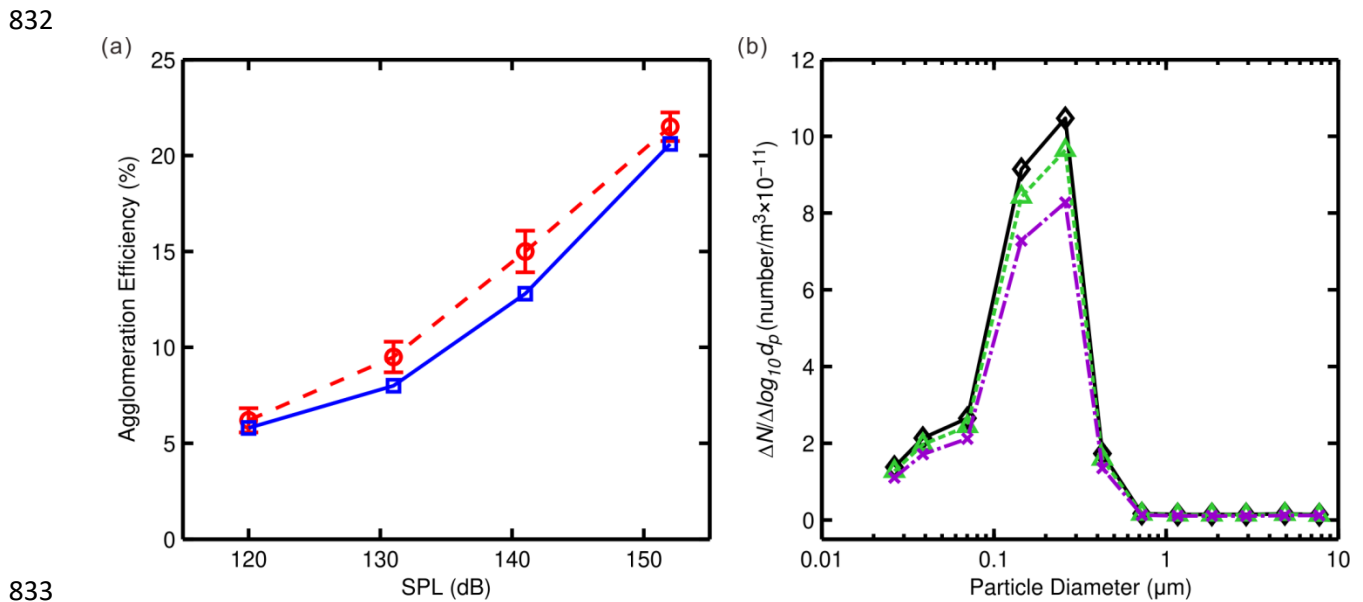
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Fig. 11. (a) Demonstration of the positions of the two straight lines along the flow direction: L1 (center line) for the dashed line and L2 (near wall) for the solid line. (b) The predicted efficiency of acoustic agglomeration along the flow direction for L1 (dashed) and L2 (solid).

819 A further investigation was conducted by varying the SPL from 120 dB to 152 dB while keeping  
 820 the frequency fixed at 1.8 kHz. The numerical results and the experimental data were  
 821 presented in Fig. 12a. It is observed that the increase of the sound pressure level can effectively  
 822 enhance the agglomeration efficiency for both numerical simulation and experiments: The  
 823 measured agglomeration efficiency in experiments increases from 6.2% at 120 dB to 21.5% at  
 824 152 dB while the predicted value increases from 5.8% at 120 dB to 20.6% at 152 dB. The  
 825 predicted agglomeration efficiency is found to be in good agreement with the experimental  
 826 data, which confirms the predictive capability of the present CFD-sectional model for the  
 827 spatially inhomogeneous coagulation/agglomeration process coupled with a fluid flow. Besides,  
 828 the predicted particle size distribution at 131 dB and 152 dB are also presented in Fig. 12b. The  
 829 profiles of the particle size distribution for two cases are similar while the decreased area under  
 830 the PSD curve with increasing SPL indicates the reduction of the number concentration of the  
 831 particles, which is in accordance with the increasing agglomeration efficiency in Fig. 12a.



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 835 Fig. 12. (a) Numerical results (square) and experimental data (circle) of the agglomeration efficiency at  
 836 varied SPL. (b) Comparison of the predicted particle size distribution at 131 dB (triangle) and 152 dB  
 837 (cross). The acoustic frequency is fixed at 1.8 kHz for all cases.  
 838

839 The particle distribution and acoustic agglomeration process shown in Figs. 9-12 demonstrates  
 840 that the present CFD-sectional model is capable to predict more details of the standing-

841 conditioned acoustic agglomeration than the conventional temporal model and the spatial  
842 inhomogeneity further confirms the rationality of multi-dimensional population balance  
843 modeling in acoustic agglomeration.

#### 844 5.2.4 Wall deposition

845 According to the Fick's law, the flux of particle deposition caused by the Brownian and  
846 turbulent diffusion is calculated by  $\vec{J} = -[(D_B + D_T)\nabla N]_{wall}$ , where  $D_B$  and  $D_T$  are the  
847 Brownian and turbulent diffusivity, respectively.

848 To quantify the effect of wall deposition, we define a coefficient of deposition loss as,

$$849 \quad DL = \dot{N}_{dep}/\dot{N}_{inlet} \times 100\% \quad (*)$$

850 where,  $\dot{N}_{dep}$  is the deposition rate of particles in total and  $\dot{N}_{inlet}$  is the inflow rate of particles  
851 entering the chamber and they are calculated by

$$852 \quad \dot{N}_{dep} = \iint_{wall} \vec{J} \cdot d\vec{S}$$

$$853 \quad \dot{N}_{inlet} = \iint_{inlet} \vec{J} \cdot d\vec{S}$$

854 The coefficient of the deposition loss of particles is presented in Fig. 13 as a function of the  
855 particle diameter. It can be seen that the maximum deposition loss occurs at the smallest  
856 particles, only accounting for 0.11% of the inflow particles. The deposition loss caused by the  
857 Brownian and turbulent diffusion is reduced with the increasing particle size and becomes  
858 negligible for particles larger than 0.2  $\mu\text{m}$  in diameter. Thus, the loss of particles due to the wall  
859 deposition is very small, and has an insignificant impact on the acoustic agglomeration in the  
860 present study.

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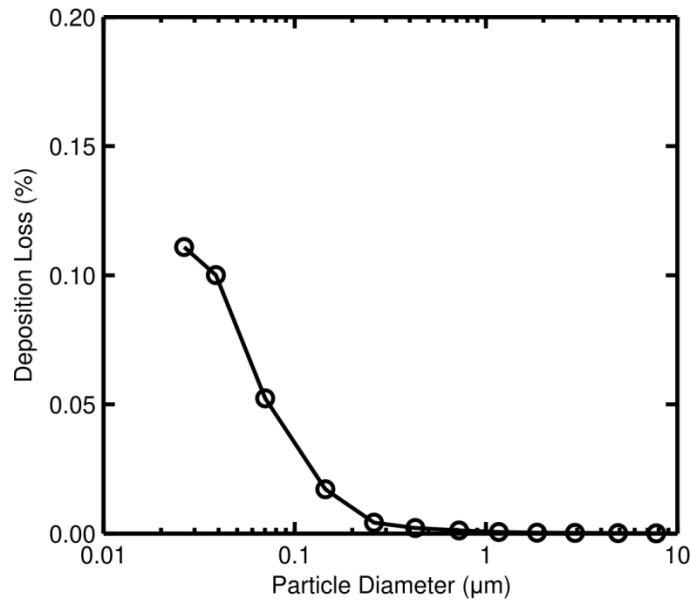


Fig. 13. The dependence of the deposition loss on the particle size at 1.8 kHz and 150 dB.

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## 6. Conclusions

867 In summary, a CFD-sectional algorithm that solves the spatially inhomogeneous population  
868 balance equation coupled with multi-dimensional flow dynamics is presented. The population  
869 balance modeling which includes convective and diffusive transport of aerosols is based on the  
870 sectional method of Gelbard and Seinfeld [15]. The solution of the section PBE is incorporated  
871 into a standard CFD framework for a laminar or turbulent flow. The following findings can be  
872 summarized from the present study:

873 1) An adaptive numerical quadrature is proposed to specify the sectional coefficients. By  
874 adaptively reducing the integration domain, the numerical quadrature achieves high  
875 computation accuracy in a convenient and economic way. The inter-equation coupling  
876 associated with the sectional PBEs is treated by the hybrid-segregated procedure to avoid the  
877 lagging effect. The source term is linearized by the operator splitting method into an explicit  
878 and an implicit part, which combines the merits of computational economy of explicit schemes  
879 and stability of implicit schemes. The solution of the sectional population balance equation is  
880 incorporated into the standard CFD simulation of incompressible flows such as the SIMPLE  
881 algorithm or PISO algorithm. Turbulence simulation can also be included to study the  
882 coagulation or diffusion in a turbulent flow.

883 2) The computation accuracy of the modified numerical quadrature with adaptive limits is  
884 shown to be greatly improved for the same number of integration nodes. The present CFD-  
885 sectional algorithm is benchmarked by an analytical solution of Brownian  
886 coagulation/agglomeration of particles in the continuum regime. The prediction of the CFD-  
887 sectional modeling is in good agreement with the self-preserving distribution. Satisfactory  
888 results are obtained concerning the prediction of the geometric mean diameter and geometric  
889 standard deviation of the particle size distribution.

890 3) The acoustic agglomeration in a standing wave field is simulated as a case study. The  
891 predicted particle size distribution agrees well with the experimental data, which demonstrates  
892 the capability of the CFD-sectional algorithm in simulating ultra-fast coagulation/agglomeration  
893 processes. On this basis, the particles are found to be distributed non-uniformly in the duct due  
894 to the combined effects of the inhomogeneous velocity field, wall deposition and acoustic  
895 agglomeration. Besides, the agglomeration efficiency is found to increase in an intermittent  
896 way along the flow direction and differ with the varied position in the cross section of the duct.

897 Although the developed CFD-sectional algorithm is limited to the phenomena of aerosol  
898 coagulation, convection, diffusion and deposition, its extension to the nucleation and  
899 condensation is conceptually straightforward. However, the present algorithm overlooks the  
900 impact of the aerosol dynamics on the mass and momentum transfer of the carrier fluids, which  
901 makes it suitable for the dilute particle-laden flows. The two-way coupling between the  
902 population balance equation and the fluid dynamics will be our goal in future.

### 903 Acknowledgements

904 This study is supported by the Republic of Singapore's Ministry of Education MOE2016-T2-1-063.

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### 1032 Figure Captions

1033

1034 Fig. 1. Illustration of the integration limits of  ${}^1\beta_{i,j,l}$  for the case of: (a)  $i = l - 1, j < l - 1$  and (b)  $i <$   
1035  $l - 1, j = l - 1$ . The black shaded area shows the “goal” region and  $v_{l-1}$  and  $v_l$  represent the lower and  
1036 upper boundaries of the agglomerating section.

1037

1038 Fig. 2. Illustration of the integration limits of  ${}^2\beta_{i,l}$  for the case of: (a)  
1039  $\theta(u + v > v_l)$  and (b)  $\theta(v_{l-1} < u + v < v_l)$ . The black shaded area shows the “goal” region and  $v_{l-1}$   
1040 and  $v_l$  represent the lower and upper boundaries of the agglomerating section.

1041

1042 Fig. 3. Illustration of the integration limits of  ${}^3\beta_{l,l}$ . The black shaded area shows the “goal” region and  
1043  $v_{l-1}$  and  $v_l$  represent the lower and upper boundaries of the agglomerating section.

1044  
1045 Fig. 4. Schematic representation of the CFD-sectional algorithm.

1046  
1047 Fig. 5. Comparison between the self-preserving particle distribution (dashed line) and the prediction of  
1048 CFD-sectional algorithm (stairs) for an aerosol of particles undergoing Brownian  
1049 coagulation/agglomeration in the continuum regime at 24 hours. The initial distribution is denoted by  
1050 the solid line.

1051  
1052 Fig. 6. Illustration of the computation domain and meshing scheme in the simulation (not to scale). (a)  
1053 Global meshing (b) Near-wall Refinement.

1054  
1055 Fig. 7. Simulation results (square) compared with experimental measurements (plus) of the particle size  
1056 distribution in a standing wave field. The initial distribution (diamond) of the aerosol at the chamber  
1057 inlet is also included.

1058  
1059 Fig. 8. (a) Spatial distribution of the total number concentration of the particles along the flow direction.  
1060 (b) Spatial distribution of the total number concentration of the particles across the flow direction. (c)  
1061 Spatial distribution of the velocity field across the flow direction.

1062  
1063 Fig. 9. (a) Demonstration of the positions of the two straight lines along the flow direction: L1 (center  
1064 line) for the dashed line and L2 (near wall) for the solid line. (b) The predicted efficiency of acoustic  
1065 agglomeration along the flow direction for L1 (dashed) and L2 (solid).  
1066